

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : LANDING/DECELERATION-LGC FMEA NO 02-1A -016 -1 REV:09/19/88

ASSEMBLY : MAIN LANDING GEAR (MLG)	CRIT. FUNC:	1
P/N RI : V070-510001	CRIT. HDW:	1
P/N VENDOR:	VEHICLE	102 103 104
QUANTITY : 8	EFFECTIVITY:	X X X
: 4 LH	PHASE(S):	PL LO OO DO LS X
: 4 RH		

PREPARED BY:	REDUNDANCY SCREEN:	A-	B-	C-
DES R. A. GORDON	APPROVED BY:	APPROVED BY (NASA):		
REL J. S. MULLEN	DES <i>R.A. Gordon 9/21/88</i>	SSM <i>Charles C. ...</i>		
QE W. J. SMITH	REL <i>W.S. ...</i>	REL <i>... 9/15</i>		
	QE <i>W.S. ...</i>	QE <i>...</i>		

ITEM:  
MAIN LANDING GEAR FITTING ASSEMBLIES, DRAG BRACE AND STRUT TRUNNIONS

FUNCTION:  
VEHICLE TO LANDING GEAR STRUT & DRAG BRACE ATTACH ASSY.

FAILURE MODE:  
STRUCTURAL FAILURE

CAUSE(S):  
OVERLOAD, DEFECTIVE PART/MATERIAL.

EFFECT(S) ON:  
(A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE

(A) LOSS OF LOAD CARRYING CAPABILITY.  
(B) DAMAGE TO VEHICLE STRUCTURE.  
(C,D) PROBABLE LOSS OF MISSION/CREW/VEHICLE DUE TO GEAR COLLAPSE.

DISPOSITION & RATIONALE:  
(A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE

(A) DESIGN  
THE FITTINGS ARE DESIGNED TO OPERATE FOR 400 CYCLES WITHOUT STRUCTURAL DEGRADATION. DESIGNED TO A MINIMUM FACTOR OF SAFETY OF 1.4 WITH STANDARD MATERIAL ALLOWABLES. MATERIALS USED ARE NOT SUSCEPTIBLE TO CORROSION DURING EXPOSURE TO EXPECTED ORBITER ENVIRONMENTS.

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(B) TEST

QUALIFICATION TEST: THE FITTINGS WERE CERTIFIED AS AN INTEGRAL PART OF THE MLG MECHANISM INSTALLATION (LANDING GEAR OPERATION) - 32 CYCLES OF THE LANDING GEAR DURING ALT, 15 DEVELOPMENT CYCLES AND 353 QUALIFICATION LIFE CYCLES FOR A TOTAL OF 400 CYCLES. (THE LANDING GEAR WAS CYCLED UP AND LOCKED TO DOWN AND LOCKED EACH TIME).

ENVIRONMENTS:

HIGH TEMP TESTS; 3 CYCLES AT 140 DEG F

COLD TEMP TESTS; 3 CYCLES AT -25 DEG F TO -40 DEG F

OMRSD: MLG WHEELWELL DETAIL INTERNAL INSPECTION; MAIN GEAR STRUT TRUNNION FITTINGS AND DRAG BRACE FITTINGS ARE INSPECTED FOR EVIDENCE OF STRUCTURAL DEGRADATION.

FREQUENCY: NINTH FLIGHT OF EVERY VEHICLE AND EVERY EIGHTH FLIGHT THEREAFTER.

(C) INSPECTION

RECEIVING INSPECTION

RECEIVING INSPECTION VERIFIES MATERIAL AND PROCESS CERTIFICATIONS.

CONTAMINATION CONTROL

CLEANLINESS AND CORROSION PROTECTION REQUIREMENTS VERIFIED BY INSPECTION

ASSEMBLY/INSTALLATION

MATERIAL USED FOR FABRICATION VERIFIED BY INSPECTION ON MANUFACTURING ORDERS. MACHINE TOLERANCES PER DRAWING AND MACHINING SPECIFICATION. INSTALLATION OF BEARING PER BEARING INSTALLATION SPECIFICATION, TOOL AIDS REQUIRED, VERIFIED BY INSPECTION. TRUNNION ASSEMBLY COMPLETE PER DRAWING AND APPLICABLE SPECIFICATIONS VERIFIED BY INSPECTION ON MANUFACTURING ORDERS.

CRITICAL PROCESSES

HEAT TREATING VERIFIED BY INSPECTION.

NONDESTRUCTIVE EVALUATION

FLUORESCENT PENETRANT INSPECTION REQUIRED AND VERIFIED BY INSPECTION.

TESTING

ACCEPTANCE TESTING IS VERIFIED BY INSPECTION.

PACKAGING/HANDLING

HANDLING AND PACKAGING REQUIREMENTS ARE VERIFIED BY INSPECTION.

(D) FAILURE HISTORY

NONE.

(E) OPERATIONAL USE

NONE.