FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CIL HARDWARE

NUMBER: 02-2A-021111 -X

SUBSYSTEM NAME: FLIGHT CONTROL MECH - RUDDER SPEED BRAKE & BF

**REVISION**: 0 02/02/88

PART DATA

PART NAME

VENDOR NAME

PART NUMBER

VENDOR NUMBER

ASSY : BODY FLAP ACTUATION

MC621-0056-0083

SRU

: SUMMER DIFFERENTIAL

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:

SUMMER DIFFERENTIAL (SECOND STAGE)

REFERENCE DESIGNATORS:

QUANTITY OF LIKE ITEMS: 2

TWO REQUIRED

## **FUNCTION:**

FIRST AND SECOND STAGE DIFFERENTIALS SUM THE RPM/TORQUE OUTPUT FROM THREE BODY FLAP HYDRAULIC MOTORS INTO A SINGLE SHAFT RPM/TORQUE OUTPUT. FIRST STAGE SUMS OUTPUT FROM TWO HYDRAULIC MOTORS. SECOND STAGE SUMS OUTPUT FROM THE FIRST STAGE DIFFERENTIAL AND THIRD HYDRAULIC MOTOR.

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08/07/98

SUBSYSTEM NAME: FLIGHT CONTROL MECH - RUDDER SPEED BRAKE & BF

**CRITICALITY OF THIS** 

LRU:

ITEM NAME: SUMMER DIFFERENTIAL

FAILURE MODE: 1R2

FAILURE MODE:

NO RPM/TORQUE OUTPUT, JAMMED DRIVELINE

MISSION PHASE:

DO DE-ORBIT

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

102 COLUMBIA

103 DISCOVERY 104 ATLANTIS

105 ENDEAVOUR

CAUSE:

I

BROKEN GEAR TOOTH/TEETH, SEIZED BEARING

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

A) PASS

B) FAIL

C) PASS

PASS/FAIL RATIONALE:

A)

B)

FAILS REDUNDANCY SCREEN "B" SINCE THE HALF-RATE EFFECT OF THE FIRST FAILURE MAY NOT BE DETECTABLE BY CREW.

C)

# - FAILURE EFFECTS -

## (A) SUBSYSTEM:

LOSS OF TWO HYDRAULIC MOTOR RPM/TORQUE INPUTS INTO SUMMER DIFFERENTIAL, CAUSING BODY FLAP TO OPERATE AT HALF-RATE.

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(B) INTERFACING SUBSYSTEM(\$):

NONE.

(C) MISSION:

NONE

(D) CREW, VEHICLE, AND ELEMENT(S):

LOSS OF MISSION, CREW/VEHICLE AFTER TWO FAILURES - DIFFERENTIAL GEAR MESH JAM AND LOSS OF THIRD HYDRAULIC MOTOR RPM/TORQUE INPUT, RESULTING IN LOSS OF BODY FLAP FUNCTION.

#### -DISPOSITION RATIONALE-

#### (A) DESIGN:

SEALED GEARBOXES ACCEPTED BY PROGRAM PER MCR 231. GEARS/SHAFTS DESIGNED BY SIZING FOR MAXIMUM TORQUE X 1.4 SAFETY FACTOR. CARBURIZED STEEL FOR GEARS VACUUM MELT PER AMS 6265 WITH CARBURIZING TO AGMA 246.01. SHOT PEEN TO MIL-S-13165. HEAVILY LOADED GEARS ARE GRIT BLASTED TO REMOVE INTERGRANULAR OXIDATION (IGO); LIGHTLY LOADED GEARS GROUND FOR IGO REMOVAL. GEAR STRESS ANALYSIS PER LEWIS EQUATION. FATIGUE ANALYSIS BASED ON GREATEST MISSION DUTY CYCLES X 4 FOR DESIGN REQUIREMENTS. BEARINGS DESIGNED FOR B-10 LIFE MINIMUM

(B) TEST:

OUALIFICATION TESTS: INCLUDES OPERATING CYCLE TEST FOR MAXIMUM LOAD CYCLES PER MISSION X 4 WITH A RUN IN AND STATIC PROOF TORQUE TO 1.5 X OPERATIONAL HYDRAULIC PRESSURE, THERMAL CYCLE -40 DEG F TO + 275 DEG F, VIBRATION FROM 20 TO 2,000 HZ RANDOM, ULTIMATE LOAD, STIFFNESS, AND FATIQUE LIFE.

ACCEPTANCE TESTS: IMPULSE AND THERMAL CYCLING, OPERATING HINGE MOMENT AND SURFACE RATE.

GROUND TURNAROUND TEST
ANY TURNAROUND CHECKOUT TESTING IS ACCOMPLISHED IN ACCORDANCE WITH OMRSD.

(C) INSPECTION:

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# RECEIVING INSPECTION

MATERIAL AND PROCESS CERTIFICATIONS VERIFIED, INCLUDING GEAR CERTIFICATIONS. CONTROLS, AND MATERIAL IDENTIFICATION, CODE, MILL SOURCE, HEAT NUMBER, CHEMICAL ANALYSIS AND HARDNESS VERIFICATION.

#### CONTAMINATION CONTROL

CONTAMINATION CONTROL PROCEDURES AND PRACTICES ARE VERIFIED.

# ASSEMBLY INSTALLATION

ASSEMBLY AND INSTALLATIONS VERIFIED BY SHOP TRAVELER MANDATORY INSPECTION POINTS (MIPS). ALIGNMENT REQUIREMENTS VERIFIED TO DRAWING REQUIREMENTS. INSPECTION VERIFIES SURFACE TEMPER INSPECTION (MICROSTRUCTURE EVALUATION WITH NITAL ETCH). SHAFT MATERIAL INSPECTED TO DRAWING REQUIREMENTS.

## NONDESTRUCTIVE EVALUATION

ULTRASONIC INSPECTION AND MAGNETIC PARTICLE INSPECTION ARE VERIFIED.

## CRITICAL PROCESSES

DRY FILM LUBRICANT, ELECTROLESS NICKEL PLATING, SHOT PEEN AND GRIT BLASTING ARE VERIFIED. HEAT TREATING, INCLUDING CARBURIZATION, IS VERIFIED.

#### **TESTING**

CERTIFICATION OF ACCEPTANCE TESTS VERIFIED.

## HANDLING/PACKAGING

HANDLING AND PACKAGING REQUIREMENTS ARE VERIFIED BY INSPECTION.

# (D) FAILURE HISTORY:

CURRENT DATA ON TEST FAILURES, FLIGHT FAILURES, UNEXPLAINED ANOMALIES, AND OTHER FAILURES EXPERIENCED DURING GROUND PROCESSING ACTIVITY CAN BE FOUND IN THE PRACA DATA BASE.

# (E) OPERATIONAL USE:

NONE.

	- APPROVALS -	<u> </u>	-
EDITORIALLY APPROVED TECHNICAL APPROVAL	: BNA - VIA APPROVAL FORM	95-CIL-009 02-24	-