

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : EPD&C - COMM. & TRACK. FMEA NO 05-6PD-22701 -1 REV: 11/06/87

ASSEMBLY : PNL 08				CRIT. FUNC: 1R
P/N RI : ME452-0102-7201				CRIT. HDW: 2
P/N VENDOR:		VEHICLE	102 103 104	
QUANTITY : 2		EFFECTIVITY:	X X X	
: TWO, 1 FOR EA ALTIMETER		PHASE(S):	PL X LO X OO	DC X LS

PREPARED BY:	DES 11/11/87 <i>W C ELDER</i>	REL 11/14/87 <i>A L MASAI</i>	QE 11/15/87 <i>J T COURSEN</i>	REDUNDANCY SCREEN: A-PASS B-PASS C-PASS
APPROVED BY:	DES <i>[Signature]</i> 11/13/87	REL <i>[Signature]</i> 11-14-87	QE <i>[Signature]</i> 11/14/87	APPROVED BY (NASA):
				SSM <i>[Signature]</i>
				REL <i>[Signature]</i> 11/20/87
				QE <i>[Signature]</i> 11/20/87
				REL EPDC <i>[Signature]</i> 11/14/87
				SSM EPDC <i>[Signature]</i> 11-20-87 <i>La B. SI</i>

ITEM:
SWITCH, TOGGLE RADAR ALTIMETER ACTIVATION.

FUNCTION:
PROVIDES POWER & CONTROL OF THE 28V DC POWER. 2/TOGGLE SWITCH, DPDT 33V73A8S4, S5.

FAILURE MODE:
FAILS OPEN, SHORT TO CASE (GROUND).

CAUSE(S):
PIECE-PART STRUCTURAL FAILURE, CONTAMINATION, VIBRATION, MECHANICAL SHOCK, PROCESSING ANOMALY.

EFFECT(S) ON:
(A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE

(A) NO EFFECT ON EPDC.

(B) LOSS OF ALTITUDE DATA IF THAT ALTIMETER SELECTED.

(C) NO EFFECT.

(D) POSSIBLE LOSS OF CREW/VEHICLE AFTER TWO SWITCH FAILURES SINCE PRECISE ALTITUDE DATA FOR CREW DETERMINATION OF SINK RATE IS REQUIRED FOR SAFE NIGHT LANDING OR LANDINGS ON RUNWAYS WITHOUT MSBLS AND TO PREVENT POSSIBLE VEHICLE DAMAGE.

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM :EPD&C - COMM. & TRACK. FMEA NO 05-6PD-22701 -1 REV:11/06/87

DISPOSITION & RATIONALE:

(A)DESIGN (B)TEST (C)INSPECTION (D)FAILURE HISTORY (E)OPERATIONAL USE

(A,B,C,D) REFER TO APPENDIX A, ITEM # 1, TOGGLE SWITCH

(B) TEST

GROUND TURNAROUND TEST - VERIFY RADAR ALTIMETER POWER ON FROM THE D&C PANEL - PERFORMED EVERY FLIGHT.

(E) OPERATIONAL USE

RADAR ALTIMETER DATA IS AVAILABLE AT 5,000 FEET. DE-ORBIT IS NOT ATTEMPTED IF CEILING IS LESS THAN 8,000 FEET (10,000 FEET IF NO MSBLS AVAILABLE) TO ENSURE GOOD VISIBILITY AT LOW ALTITUDE. MOST ORBITER RUNWAYS ARE EQUIPPED WITH MSBLS GROUND STATIONS WHICH PROVIDE A REDUNDANT SOURCE OF LOW ALTITUDE DATA DOWN TO 50 FEET. RADAR ALTIMETER DATA IS DISPLAYED ON THE HUD AND AVVI NEXT TO NAVIGATION ALTITUDE DATA. CREW CAN ISOLATE A FAILED RADAR ALTIMETER AND THEN SELECT THE OTHER ALTIMETER, IF AVAILABLE, OR DISREGARD RADAR ALTIMETER DATA AND RELY ON NAVIGATION ALTITUDE AND/OR VISUAL CUES.