

## FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL HARDWARE

NUMBER: 05-6WB-1001-X

SUBSYSTEM NAME: EP0&amp;C - ECLSS - ATCS/ABS

REVISION : 4 07/03/91

	PART NAME VENDOR NAME	PART NUMBER VENDOR NUMBER
LRU :	PANEL L1A2	V070-730271
SRU :	SWITCH, TOGGLE	ME452-0102-7354

## PART DATA

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:  
SWITCH, TOGGLE. NH3 CONTROLLERS.

REFERENCE DESIGNATORS: 31V73A1A2 542  
: 31V73A1A2 543

QUANTITY OF LIKE ITEMS: 2  
(TWO), ONE PER CONTROLLER

FUNCTION:  
ACTIVATES NH3 FLOW CONTROL "A" OR "B" CIRCUITRY IN THE PRIMARY/GPC OR  
SECONDARY POSITIONS.

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE  
 NUMBER: 05-6WB-1001-03

SUBSYSTEM: EPD&C - ECLSS - ATCS/ABS  
 LRU :PANEL LIA2  
 ITEM NAME: SWITCH, TOGGLE

REVISION# 4 07/03/91 R

CRITICALITY OF THIS  
 FAILURE MODE:1/1

FAILURE MODE:  
 FAILS CLOSED IN THE "SECONDARY" POSITION, PREMATURE CLOSURE, POLE-TO-  
 POLE SHORT.

MISSION PHASE:  
 00 ON-ORBIT  
 00 DE-ORBIT

■ VEHICLE/PAYLOAD/KIT EFFECTIVITY: 102 COLUMBIA  
 : 103 DISCOVERY  
 : 104 ATLANTIS  
 : 105 ENDEAVOUR

CAUSE:  
 PIECE PART STRUCTURAL FAILURE, CONTAMINATION, VIBRATION, MECHANICAL  
 SHOCK, PROCESSING ANOMALY

CRITICALITY 1/1 DURING INTACT ABORT ONLY? N

REDUNDANCY SCREEN A) N/A  
 ■ B) N/A  
 ■ C) N/A

PASS/FAIL RATIONALE:  
 A)  
 B)  
 C)

MASTER MEAS. LIST NUMBERS: V63S1190E  
 : V63S1182E  
 : V63S1186E  
 : V63S1178E

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- FAILURE EFFECTS -  
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(A) SUBSYSTEM:

ACTIVATES ONE AMMONIA BOILER THROUGH SECONDARY CONTROLLER.

(B) INTERFACING SUBSYSTEM(S):

ADDITIONAL HEAT REMOVAL FROM FREON COOLANT LOOPS PROVIDED BY ONE AMMONIA SYSTEM. POSSIBLE OVERCOOLING OF THE FREON/H<sub>2</sub>O HEAT INTERCHANGER.

(C) MISSION:

POSSIBLE LOSS OF MISSION OBJECTIVES DUE TO LOSS OF PAYLOAD POSTLANDING COOLING.

■ (D) CREW, VEHICLE, AND ELEMENT(S):

FIRST FAILURE CAN RESULT IN LOSS OF CREW/VEHICLE DUE TO LOSS OF ORBITER COOLING IF WORKAROUND PROCEDURE IS NOT IMPLEMENTED IN TIME. EXCESSIVE LOWERING OF FREON LOOPS' TEMPERATURE CAN FREEZE THE FREON/H<sub>2</sub>O INTERCHANGER WHICH WILL RESULT IN RUPTURE OF WATER AND FREON LOOPS.

■ (E) FUNCTIONAL CRITICALITY EFFECTS:

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- DISPOSITION RATIONALE -  
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(A) DESIGN:

REFER TO APPENDIX A, ITEM NO. 1 - TOGGLE SWITCH.

■ (B) TEST:

REFER TO APPENDIX A, ITEM NO. 1 - TOGGLE SWITCH.  
THIS FAILURE MODE IS TESTED FOR WHENEVER THE VEHICLE IS POWERED UP (INFLIGHT AND DURING GROUND TURNAROUND) BY VERIFYING THAT THE NH<sub>3</sub> BOILER DOES NOT COME ON SPONTANEOUSLY.

(C) INSPECTION:

REFER TO APPENDIX A, ITEM NO. 1 - TOGGLE SWITCH.

(D) FAILURE HISTORY:

REFER TO APPENDIX A, ITEM NO. 1 - TOGGLE SWITCH.

■ (E) OPERATIONAL USE:

FAILURE IS INDICATED BY ONBOARD ALARM - "EVAP OUT TEMPERATURE". FOR SECONDARY AMMONIA CONTROLLER FAILED ON, ACTIVATE REDUNDANT AIR WATER LOOP. IF RADIATOR FLOW ACTIVATED, POSITION RADIATOR OUT TEMPERATURE SWITCH TO HIGH SET POINT UNTIL NH<sub>3</sub> DEPLETION, OR TURN OFF A FREON PUMP TO PREVENT FREEZING OF INTERCHANGER.

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- APPROVALS -

RELIABILITY ENGINEERING:	D. ANVARI	:	<i>D.A. Mevas</i>	<i>CL the 2-11-91</i>
DESIGN ENGINEERING	: J. L. PECK	:	<i>[Signature]</i>	<i>1/9/91</i>
DESIGN MANAGER	: G. M. ANDERSON	:	<i>[Signature]</i>	<i>1/12/91</i>
QUALITY MANAGER	: R. M. SPURLOCK	:	<i>[Signature]</i>	<i>7/12/91</i>
NASA RELIABILITY	:	:	<i>[Signature]</i>	<i>1/8/92</i>
NASA SUBSYSTEM MANAGER	:	:	<i>[Signature]</i>	<i>1/12/92</i>
NASA EPD&C RELIABILITY	:	:	<i>[Signature]</i>	<i>1-17-92</i>
NASA QUALITY ASSURANCE	:	:	<i>[Signature]</i>	<i>for R.J. Santora</i>
NASA EPD&C SUBSYS MGR	:	:	<i>[Signature]</i>	<i>for F. Alaris 1/30/92</i>