

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : EPD&C/PAYLOAD INTERFACE FMEA NO 05-6X -233JA -1 REV:12, 23 83

ASSEMBLY : PANEL RL1
P/N RI : RWR80S1211FR CRIT. FUNC: 1B
P/N VENDOR: CRIT. EDW: 2
QUANTITY : 2 VEHICLE 102 103 104
EFFECTIVITY: X X X
: TWO, PAYLOAD AUXILIARY PHASE(S): PL LO OO X OO LS
: CONTROL CIRCUIT

REUNDANCY SCREEN: A-PASS B-PASS C-PASS
PREPARED BY: APPROVED BY: APPROVED BY (NASA):
DES R PHILLIPS DES *[Signature]* SSM *[Signature]*
REL T KIMURA REL *[Signature]*
QE J. COURSEN QE *[Signature]* SPOC *[Signature]*
SSM *[Signature]*

ITEM:
RESISTOR, CURRENT LIMITING, 1.2K, 2W - PAYLOAD AUXILIARY POWER CONTROL CIRCUIT

FUNCTION:
UPON MANUAL COMMAND FROM THE PAYLOAD AUXILIARY POWER SWITCH, LIMITS CURRENT TO THE RPC CONTROL CIRCUIT. J2V73A1A1A1R1, J2V73A1A1A1R2

FAILURE MODE:
OPEN

CAUSE(S):
STRUCTURAL FAILURE (MECHANICAL STRESS, VIBRATION), ELECTRICAL STRESS, THERMAL STRESS, PROCESSING ANOMALY

EFFECT(S) ON:
(A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE (E) FUNCTIONAL CRITICALITY EFFECT:

(A) LOSS OF MAIN DC BUS A OR B POWER TO ITS ASSOCIATED PAYLOAD AUXILIARY BUS.

(B) LOSS OF REDUNDANCY. THE PAYLOAD REMAINS POWERED BY ITS REDUNDANT SOURCE.

(C) FIRST FAILURE - NO EFFECT.

(D,E) POSSIBLE LOSS OF CREW/VEHICLE WHEN SECOND FAILURE (LOSS OF SECOND SOURCE) OCCURS DURING NORMAL PAYLOAD OPERATIONS.

NOTE: FAILURE EFFECTS AND CRITICALITY OF RESISTOR ARE USER DEPENDENT. EFFECTS AND CRITICALITY WILL CHANGE ON A FLIGHT-BY-FLIGHT BASIS AND ARE DEPENDENT UPON THE PAYLOAD AND THE METHOD IN WHICH THE PAYLOAD WIRING IS DESIGNED TO TAKE ADVANTAGE OF THE REDUNDANCY OF PAYLOAD POWER CIRCUITS AVAILABLE ON THE ORBITER.

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM : EPD&C/PAYLOAD INTERFACE FMEA NO 05-6X -2330A -1 REV:12 20 81

THE WORSE-CASE CRITICALITY WILL BE 1R/2 FOR HAZARDOUS AND/OR CLASSIFIED PAYLOADS WHICH MAY BE DOWNGRADED TO 1R/3 WITH THE APPROPRIATE PAYLOAD POWER UTILIZATION WIRING DESIGN.

DISPOSITION & RATIONALE:

(A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE:

(A, B, C, D) DISPOSITION AND RATIONALE:

REFER TO APPENDIX E, ITEM NO. 3 - RESISTOR, WIRE WOUND

(B) GROUND TURNAROUND TEST

VERIFY MAIN DC BUS A AND B VOLTAGE AT PAYLOAD STATION DISTRIBUTION PANEL AND PAYLOAD INTERFACE. CYCLE PAYLOAD AUXILIARY ON/OFF SWITCH AND MONITOR STIMULI COMMANDS, DISCRETE EVENTS, AND BUS VOLTAGE. APPLICABLE FOR ALL FLIGHTS CONTINGENT ON MISSION REQUIREMENTS.

(E) OPERATIONAL USE

FOR SOME PAYLOADS, CONTINGENCY IN-FLIGHT MAINTENANCE AND EXTRA-VEHICULAR ACTIVITY PROCEDURES HAVE BEEN DEVELOPED TO SUPPORT CRITICAL FUNCTIONS. e.g. PAYLOAD RE-STOW OR DEPLOY COMPLETION, IF SWITCH FAILS.