

## FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL HARDWARE

NUMBER: 06-1B-0750-X

SUBSYSTEM NAME: ARS - COOLING

REVISION : 7 06/26/92

	PART NAME VENDOR NAME	PART NUMBER VENDOR NUMBER
■ LRU :	REGENERABLE CO2 REMOVAL SYSTEM	MC623-0016
■ SRU :	VALVE, MANUAL	SV807021

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 PART DATA
 

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■ EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:  
FLOW CONTROL VALVE

■ QUANTITY OF LIKE ITEMS: 1

■ FUNCTION:

THE VALVE IS SET BEFORE LAUNCH BASED ON CREW SIZE. THIS VALVE MODULATES THE RCRS PROCESS AIR FLOW TO MAINTAIN PROPER CABIN PPO2 AND RELATIVE HUMIDITY.

FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CRITICAL FAILURE MODE  
NUMBER: 06-1B-0750-03

SUBSYSTEM: ARS - COOLING  
LRU :REGENERABLE CO2 REMOVAL SYSTEM  
ITEM NAME: VALVE, MANUAL

REVISION# 7 06/26/92 R

CRITICALITY OF THIS  
FAILURE MODE:2/2

■ FAILURE MODE:  
EXTERNAL LEAKAGE

MISSION PHASE:  
00 ON-ORBIT

■ VEHICLE/PAYLOAD/KIT EFFECTIVITY: 102 COLUMBIA  
: 105 ENDEAVOUR

■ CAUSE:  
MECHANICAL SHOCK, VIBRATION, CORROSION

■ CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

■ REDUNDANCY SCREEN A) N/A  
B) N/A  
C) N/A

PASS/FAIL RATIONALE:

- A)
- B)
- C)

- FAILURE EFFECTS -

- (A) SUBSYSTEM:  
POTENTIAL SYSTEM CONTAMINATION DUE TO PROCESSING OF UNFILTERED AIR  
(FILTER IS UPSTREAM OF VALVE). POSSIBLE LOSS OF USE OF THE RCRS.
- (B) INTERFACING SUBSYSTEM(S):  
LOSS OF USE OF THE RCRS WILL RESULT IN HIGH CABIN PPO2. POSSIBLE  
REDUCED COOLING AIR FLOW THROUGH FLIGHT DECK AVIONICS LRU'S.
- (C) MISSION:  
POSSIBLE EARLY MISSION TERMINATION. DECISION BASED ON MAGNITUDE OF

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LEAK.

- (D) CREW, VEHICLE, AND ELEMENT(S):  
NO EFFECT. EARLY MISSION TERMINATION WILL PRECLUDE LOSS OF CREW OR VEHICLE.
- (E) FUNCTIONAL CRITICALITY EFFECTS:  
N/A

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- DISPOSITION RATIONALE -  
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- (A) DESIGN:  
THE VALVE IS A MANUALLY ACTUATED ALUMINUM BUTTERFLY VALVE. VALVE HAS TWO FLOW POSITION SETTINGS WITH A POSITIVE LOCK NUT TO ELIMINATE ANY POTENTIAL VALVE MOVEMENT. HOUSING IS CONSTRUCTED FROM MACHINED ALUMINUM STOCK.
- (B) TEST:  
QUALIFICATION TEST FOR 100 MISSIONS:  
VALVE IS QUAL TESTED WHILE IS INSTALLED AT THE RCRS ASSEMBLY LEVEL. RANDOM VIBRATION INCREASING AT PLUS 6 db/oct FROM 20 TO 45 HZ; CONSTANT AT 0.003 g<sup>2</sup>/HZ FROM 45 TO 1000 HZ; DECREASING AT MINUS 6 db/oct FROM 1000 TO 2000 HZ FOR 48 MINUTES PER AXIS IN THREE ORTHOGONAL AXES.  
ACCEPTANCE TEST:  
PROOF PRESSURE TESTED AT 0.5 PSID INTERNALLY APPLIED, WITH THE BUTTERFLY FULLY OPEN, WITHOUT PERMANENT DEFORMATION OR DEGRADING THE PERFORMANCE. FLOW TEST IS PERFORMED DURING ACCEPTANCE TESTING TO VERIFY NO FLOW RESTRICTION. VALVE WAS CYCLED FOR 300 TIMES TO VERIFY THE DESIGN IS CAPABLE OF MEETING LIFE REQUIREMENT.  
OMRSD:  
ANY TURNAROUND CHECKOUT TESTING IS ACCOMPLISHED IN ACCORDANCE WITH OMRSD AT SYSTEM LEVEL.
- (C) INSPECTION:  
RECEIVING INSPECTION  
INCOMING PART/MATERIAL IDENTIFICATION AND CERTIFICATION VERIFIED BY INSPECTION. VALVE ATP VERIFIED AT VENDOR BY H. S. SOURCE INSPECTION.  
  
CONTAMINATION CONTROL  
CONTAMINATION CONTROL PROCESSES AND CLEAN AREAS VERIFIED BY INSPECTION. VALVE CLEAN REQUIREMENTS VERIFIED BY INSPECTION.  
  
ASSEMBLY/INSTALLATION  
DIMENSIONAL VERIFICATION PERFORMED AT VENDOR BY INSPECTION. POSITIVE LOCK-NUT POSITION VERIFICATION BY INSPECTION. ASSEMBLY AND INSTALLATION VERIFIED BY INSPECTION.

