

SHUTTLE CRITICAL ITEMS LIST - ORBITER

SUBSYSTEM :ACTUATION MECH-PBD FMEA NO 02-4B -202 -2 REV:03/08/88

ASSEMBLY :PBD ACTUATION			CRIT. FUNC:	1
P/N RI :MC162-0008-0004, -0034			CRIT. HDW:	1
P/N VENDOR:181650-1 CURTISS-WRIGHT	VEHICLE	102	103	104
QUANTITY :2	EFFECTIVITY:	X	X	X
	PHASE(S):	PL	LO	OO X DO LS

		REDUNDANCY SCREEN:	A-	B-	C-
PREPARED BY:		APPROVED BY:	APPROVED BY (NASA):		
DES M. A. ALLEN		DES <i>D. Campbell</i>	SSM <i>R.C. Moore</i>	<i>3/18/88</i>	
REL M. B. MOSKOWITZ		REL <i>M.B. Moskowitz</i>	REL <i>M.B. Moskowitz</i>		
QE W. J. SMITH		QE <i>W.J. Smith</i>	QE <i>W.J. Smith</i>		

ITEM:
GEARBOX, POWER DRIVE UNIT

FUNCTION:
MECHANICAL POWER TRANSMISSION (GEARBOX) WHICH DISTRIBUTES POWER FROM THE POWER SOURCE, PROVIDES THE ROTARY MOTION TO DRIVE THE OUTPUT TORQUE SHAFTS TO DEPLOY THE PAYLOAD BAY DOORS TO POSITION.

FAILURE MODE:
FAILS FREE

CAUSE(S):
SLIPS AT LESS THAN MINIMUM ALLOWABLE TORQUE, FAILURE/DEFLECTION OF INTERNAL PART, FATIGUE, VIBRATION

EFFECTS ON:
(A) SUBSYSTEM (B) INTERFACES (C) MISSION (D) CREW/VEHICLE
(A,B) LOSS OF CAPABILITY TO OPEN OR CLOSE PAYLOAD BAY DOOR.
(C) LOSS OF MISSION IF PAYLOAD BAY DOOR CANNOT BE OPENED.
(D) LOSS OF CREW/VEHICLE IF DOOR CANNOT BE CLOSED.

DISPOSITION & RATIONALE:
(A) DESIGN (B) TEST (C) INSPECTION (D) FAILURE HISTORY (E) OPERATIONAL USE

(A) DESIGN
GEARS ARE DESIGNED WITH HIGH MARGINS. MAXIMUM CALCULATED TOOTH BENDING STRESS APPROXIMATELY 90,000 PSI, ULTIMATE ALLOWABLE 270,000 PSI. THE COMPONENTS OF THE ACTUATOR ASSEMBLY ARE FABRICATED OF CPS 4907 STEEL HEAT TREATED TO A HARDNESS OF ROCKWELL C54-56. BEARINGS INCORPORATE MULTIPLE ROTATING SURFACES. DESIGN OF THE ACTUATION SYSTEM PERMITS PARTIAL WORKAROUND OF THIS FAILURE MODE BY EXTRAVEHICULAR ACTIVITY (EVA) CREW IF PAYLOAD DOES NOT LIMIT ACCESS.

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(B) TEST

QUALIFICATION TESTS: THE QUALIFICATION DRIVE SYSTEM IS CERTIFIED PER CR-29-162-0008-0001E AND THE QUALIFICATION ACTUATION MECHANISM INSTALLATION CERTIFIED PER CR-29-594125-001G. THE DRIVE SYSTEM QUALIFICATION TEST INCLUDES: HUMIDITY TEST - PER MIL-STD-810B METHOD 507 PROCEDURE IV, CYCLE PDU DURING SECOND AND FOURTH HUMIDITY CYCLE; QUALIFICATION VIBRATION TEST (QAVT) - 20 TO 2,000 HZ RANGE WITH MAXIMUM OF 0.067 g^2 /HZ FROM 80 TO 350 HZ FOR 2 1/2 MINUTES/AXIS IN ACCORDANCE WITH SP-T-0023B, MONITOR ELECTRICAL CONTINUITY DURING VIBRATION; FLIGHT VIBRATION - 20 TO 2,000 HZ RANGE WITH MAX OF 0.03 g^2 /HZ FROM 100 TO 250 HZ FOR 4.5 MINUTES/AXIS LEVEL "A", AND 0.008 g^2 /HZ FROM 100 TO 250 HZ FOR 94 MINUTES/AXIS LEVEL "B"; SHOCK TEST - BASIC DESIGN SHOCK PER MIL-STD-810B METHOD 516.1, PROCEDURE I AND TRANSIENT SHOCK AT 5 TO 35 HZ +/- 0.25 g PEAK.

QUAL TESTS ALSO INCLUDE: THERMAL VACUUM TEST - THERMALLY CYCLED 5 TIMES BETWEEN -100 DEG F AND +157 DEG F AT A VACUUM OF 1 X 10⁻⁶ TORR FOR 55 HOURS, DRIVE SYSTEM CYCLED AT EACH -65 DEG F AND +157 DEG F; THERMAL CYCLING TEST - CYCLED 5 TIMES BETWEEN -100 DEG F AND +340 DEG F WITH DRIVE SYSTEM CYCLED AT EACH -100 DEG F AND +157 DEG F WITH 60 MINUTES DWELL AT EACH TEMPERATURE EXTREME; OPERATING LIFE TEST - DRIVE SYSTEM CYCLED 1,800 TIMES AT ROOM TEMPERATURE, INCLUDES MOTOR 1 AND 2 CYCLED 150 TIMES EACH INDIVIDUALLY WITHIN 126 SECONDS/STROKE AND 1,500 TIMES WITH BOTH MOTORS DRIVING TOGETHER WITHIN 63 SECONDS/STROKE; MECHANICAL STOPS TEST - 100 TIMES WITH BOTH MOTORS INTO HARD STOP IN EACH DIRECTION AT NO LOADS; STIFFNESS TEST - MEASURED SPRING RATE OF ROTARY ACTUATOR, TORQUE SHAFT, PDU AND SHAFT HANGER - GREATER THAN 0.5 X 10⁻⁶ INCH-LB/RADIAN AT ROTARY ACTUATOR ARM); POWER CONSUMPTION TEST, IRREVERSIBILITY TEST, FREEPLAY TEST WAS CONDUCTED AS DEFINED IN THE ACCEPTANCE TESTS.

CERTIFICATION BY ANALYSIS/SIMILARITY INCLUDED: FUNGUS, OZONE, PACKAGING, LIMIT/ULTIMATE LOAD, ACCELERATION, LANDING SHOCK, SYSTEM STIFFNESS, TEMPERATURES, HUMIDITY, LIFE, PRESSURE, SHOCK AND VIBRATION FOR ITEMS OF DRIVE SYSTEM NOT TESTED. THE PBD ACTUATING MECHANISM INSTALLATION WAS SUBJECTED TO SYSTEM QUALIFICATION TESTS ON THE 15 FOOT PBD TEST ARTICLES (087) AND ON A 60 FOOT TEST RIG, TESTS INCLUDED: ACCEPTANCE - TO CONFIRM ALL COMPONENTS HAVE BEEN ASSEMBLED AND RIGGED PER MLO308-0032 ON THREE TEST SPECIMENS.

TESTS ON FORWARD 15 FOOT PANEL INCLUDED: ORBITAL FUNCTIONS - 3 THERMAL CONDITIONS WITH SIMULATED THERMAL DISTORTIONS OF THE FORWARD BULKHEAD AND THE LH AND RH LONGERON SILLS, 2 MOTOR OPERATIONS 22 CYCLES AT LESS THAN 63 SECONDS PER STROKE, EACH MOTOR OPERATION 3 CYCLES AT LESS THAN 126 SECONDS/STROKE FOR EACH OF THE 3 ORBITAL FUNCTIONS; OPERATING LIFE TESTS. - PBD MECHANISM CYCLED 93 TIMES, 55 TIMES WITH TWO MOTOR OPERATION AND 19 TIMES WITH SINGLE MOTOR OPERATIONS; ACOUSTIC TEST - 25 HZ TO 8,000 HZ FOR 5 MINUTES.

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TESTS ON AFT 15 FOOT PANEL INCLUDED: ORBITAL FUNCTIONS - 3 THERMAL CONDITIONS WITH SIMULATED THERMAL DISTORTIONS OF THE AFT BULKHEAD AND THE LH AND RH LONGERON SILL, 2 MOTOR OPERATIONS 27 CYCLES AT LESS THAN 63 SECONDS/STROKE, EACH MOTOR OPERATION 3 CYCLES AT LESS THAN 126 SECONDS/STROKE FOR EACH OF THE 3 ORBITAL FUNCTIONS; OPERATING LIFE TESTS - MECHANISM CYCLED 243 TIMES 205 TIMES WITH TWO MOTOR OPERATION AND 19 TIMES WITH SINGLE MOTOR OPERATIONS; ACOUSTIC TEST - 25 HZ TO 6,000 HZ FOR 5 MINUTES.

TESTS ON 60 FOOT TEST RIG INCLUDED: AMBIENT CYCLING - 3 THERMAL CONDITIONS WITH THERMAL DISTORTIONS OF THE SILL LONGERON, 2 MOTOR OPERATIONS 22 CYCLES AT LESS THAN 63 SECONDS/STROKE; 10 WITHOUT DISTORTION SIMULATION; EACH SINGLE MOTOR OPERATION 3 WITHOUT DISTORTION AND 6 WITH DISTORTION AT LESS THAN 126 SECONDS/STROKE; TEMPERATURE CYCLING - 3 THERMAL CONDITIONS WITH THERMAL DISTORTIONS OF THE SILL LONGERON, 2 MOTOR OPERATIONS 22 CYCLES AT -42 DEG F AT LESS THAN 63 SECONDS/STROKE; 10 WITHOUT DISTORTION SIMULATION, EACH SINGLE MOTOR OPERATION 3 WITHOUT DISTORTION SIMULATION AND WITH DISTORTION AT LESS THAN 126 SECONDS/STROKE AT -42 DEG F; ORBITAL FUNCTIONAL TEST - REPEAT THE ABOVE AMBIENT AND -42 DEG F TESTS FOR A TOTAL OF 80 MECHANISM CYCLES. CERTIFICATION BY ANALYSIS/SIMILARITY INCLUDES FUNGUS, OZONE, PACKAGING, THERMAL/VACUUM, SALT SPRAY, ACOUSTIC, SHOCK, LIMIT/ULTIMATE LOADS, ACCELERATION, SAND/DUST AND MARGIN OF SAFETY.

ACCEPTANCE TEST: TESTS ON THE MC162-0008 COMPONENTS INCLUDES: EXAMINATION OF PRODUCT - WEIGHT, WORKMANSHIP, DIMENSIONS, CONSTRUCTION, CLEANLINESS, FINISH, IDENTIFICATION, MARKINGS, TRACEABILITY AND USE OF CERTIFIED MATERIALS AND PROCESSES; ACCEPTANCE VIBRATION (AVT) - 20 TO 2,000 HZ RANGE WITH MAXIMUM OF 0.04 g²/HZ FROM 80 TO 350 HZ FOR 30 SECONDS/AXIS MINIMUM, ELECTRICAL CONTINUITY MONITORED DURING TESTS AND PDU CYCLED BEFORE AND AFTER VIBRATION; ACCEPTANCE THERMAL TEST (ATT) - THERMALLY CYCLED FROM +70 DEG F TO +320 DEG F, TO +157 DEG F, TO -80 DEG F, TO +320 DEG F, TO +157 DEG F, TO 70 DEG F WITH CONTINUITY MONITORED THROUGHOUT, PDU WAS CYCLED 6 TIMES AT EACH +157 DEG F AND 6 TIMES AT -80 DEG F AT LESS THAN 63 SECONDS/STROKE TWO MOTOR OPERATIONS AND 126 SECONDS/STROKE SINGLE MOTOR OPERATIONS; POWER CONSUMPTION TESTS -INPUT POWER MAX OF 450 WATTS/MOTOR AT RATED LOAD AND SINGLE MOTOR TIME OF 126 SECONDS/STROKE AND 63 SECONDS/STROKE DUAL MOTOR OPERATION; INSULATION RESISTANCE, DIELECTRIC STRENGTH - PER MF0004-002 EXCEPT TEST VOLTAGE WAS 750 VRMS. CYCLING TEST - SINGLE MOTOR DRIVE 20 CYCLES EACH AND DUAL MOTORS 40 CYCLES. FREEPLAY TEST - MAXIMUM OF 1.0 DEGREES WITH 10 IN LBS REVERSING TORQUE ON EACH ACTUATOR; ACTUATOR STALL - CONTINUOUS STALL FOR 126 SECONDS AT FULL INVERTOR POWER 120 VOLTS AC.

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ACCEPTANCE TESTS ALSO INCLUDE: IRREVERSIBILITY TEST - 650 INCH-LB AT PDU OUTPUT SHAFT; TRAVEL LIMIT TEST - ACTUATOR STOPPED BY LIMIT SWITCHES AND BY HARD STOPS WITH SWITCHES DEENERGIZED; MANUAL OPERATIONS - LESS THAN 100 INCH-LB TORQUE TO ENGAGE AND DISENGAGE THE PDU; BACKDRIVE - FORCE AT ACTUATOR OUTPUT ARM ROD ATTACH POINT LESS THAN 150 POUNDS; TORQUE LIMITS - ROTARY ACTUATOR OUTPUT LIMITS AT ROOM AMBIENT ARE 6,200 TO 8,950 INCH-LB, OUTPUT LIMITS AT +157 DEG F AND -65 DEG F ARE 6,200 TO 10,000 INCH-LB; PROOF LOAD TESTS - 3/4 O.D. DRIVE SHAFTS TESTED TO 325 INCH-LB AND 1.0 O.D. DRIVE SHAFTS TESTED TO 650 INCH-LB; FRICTION TORQUE TEST - TORQUE ON SHAFT SUPPORT IS LESS THAN 1.0 INCH-LB. EACH TORQUE SHAFT IS PROOF LOADED DURING ACCEPTANCE.

OMRSD: GROUND TURNAROUND INCLUDES VISUAL INSPECTION OF HARDWARE TO INSURE THAT PARTS ARE NOT BROKEN OR DEFORMED AND MONITORING FUNCTIONAL TEST FOR EVIDENCE OF BINDING OR JAMMING. PROPER FUNCTION OF THE COMPONENTS IS VERIFIED PERIODICALLY AS PART OF THE MAINTENANCE SAMPLING PROGRAM.

(C) INSPECTION

RECEIVING INSPECTION

STEEL PER CPS 4907 VERIFIED BY RECEIVING INSPECTION. ALL PURCHASED PARTS DATA PACKAGES INSPECTED BY RECEIVING INSPECTION.

CONTAMINATION CONTROL

BEARINGS INSPECTED PRIOR TO INSTALLATION FOR CONTAMINATION. DETAIL PARTS ARE CLEANED TO A 300 LEVEL. SUPPLIER CONTAMINATION CONTROL AND CORROSION PROTECTION VERIFIED BY INSPECTION.

ASSEMBLY/INSTALLATION

GEARS ARE SHOT PEENED PER MIL-S-13165 TO PRECLUDE FATIGUE. BEARING LUBRICATION AND SEAL INSTALLATION VERIFIED. ALL MACHINED PARTS ARE DEBURRED AND VERIFIED PER DRAWING. INSTALLATION PROCEDURE VERIFIED BY INSPECTION. DCAS MANDATORY INSPECTION POINTS (MIPS) IMPOSED ON MANUFACTURING, INSTALLATION, AND ASSEMBLY. POWER DRIVE UNIT (PDU) ASSEMBLY PER CPS 6916 VERIFIED BY INSPECTION. DIMENSIONS AND GEAR TEETH SURFACE ROUGHNESS VERIFIED BY INSPECTION.

NONDESTRUCTIVE EVALUATION

MAGNETIC PARTICLE INSPECTION OF GEARS PER MIL-I-6868 VERIFIED BY INSPECTION.

CRITICAL PROCESSES

HEAT TREAT OF CPS 4907 STEEL GEARS PER CPS 7019 VERIFIED BY INSPECTION. VACUUM DEPOSITED CADMIUM PLATING OF GEARS PER MIL-C-8837, TYPE II, CLASS 2 VERIFIED BY INSPECTION.

TESTING

ROCKWELL HARDNESS VERIFIED ON GEARS. ATP OF PDU PER CPS 6884 VERIFIED BY INSPECTION.

HANDLING/PACKAGING

HANDLING AND PACKAGING REQUIREMENTS ARE VERIFIED.

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(D) FAILURE HISTORY

THERE HAVE BEEN NO ACCEPTANCE TEST, QUALIFICATION TEST, FIELD OR FLIGHT FAILURES ASSOCIATED WITH THIS FAILURE MODE.

(E) OPERATIONAL USE

EVA WORKAROUND IS POSSIBLE TO CLOSE DOOR IF PAYLOAD DOES NOT LIMIT ACCESS.