

**FAILURE MODES EFFECTS ANALYSIS (FMEA) – NON-CIL HARDWARE  
NUMBER: 05-6-2756 -X**

**SUBSYSTEM NAME: ELECTRICAL POWER DISTRIBUTION & CONTROL**  
**REVISION: 1** 07/26/98

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**PART DATA**

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	<b>PART NAME</b>	<b>PART NUMBER</b>
	<b>VENDOR NAME</b>	<b>VENDOR NUMBER</b>
LRU	: MID MCA-4	V070-764500
LRU	: MID MCA-4	V070-764640
SRU	: RELAY, GENERAL PURPOSE	MC455-0129-0001

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**EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:**  
RELAY, GENERAL PURPOSE, 4 POLE - MID MCA 4 THREE-PHASE PLBM AC BUS 2

**REFERENCE DESIGNATORS:** 40V76A120K49  
40V76A120K61

**QUANTITY OF LIKE ITEMS:** 2  
TWO

**FUNCTION:**  
UPON CREW INITIATED SWITCH COMMANDS, THE CONTACTS OF TWO SERIES RELAYS CONNECT MID MOTOR CONTROL ASSEMBLY #4 AC BUS AC2 (PHASE A, B, AND C) TO PAYLOAD BAY MECHANICAL (PLBM) AC BUS 2 FOR FREON RADIATOR LATCH, REMOTE MANIPULATOR LATCH AND KU-BAND DEPLOY/STOW MOTORS.

## FAILURE MODES EFFECTS ANALYSIS FMEA - NON-CIL FAILURE MODE

NUMBER: 05-6-2756-01

REVISION#: 1 07/26/99

SUBSYSTEM NAME: ELECTRICAL POWER DISTRIBUTION &amp; CONTROL

LRU: MID MCA-4

CRITICALITY OF THIS

ITEM NAME: RELAY, GENERAL PURPOSE

FAILURE MODE: 1R3

## FAILURE MODE:

OPEN, FAILS TO CONDUCT, FAILS TO TRANSFER (TO CLOSE), INADVERTENTLY OPENS,  
SHORTS TO GROUND (COIL)MISSION PHASE: OO ON-ORBIT  
DO DE-ORBITVEHICLE/PAYLOAD/KIT EFFECTIVITY: 102 COLUMBIA  
103 DISCOVERY  
104 ATLANTIS  
105 ENDEAVOUR

## CAUSE:

CONTAMINATION, PIECE PART FAILURE, VIBRATION, MECHANICAL SHOCK, THERMAL  
STRESS, PROCESSING ANOMALY

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN A) PASS  
B) PASS  
C) PASS

## PASS/FAIL RATIONALE:

A)

B)

C)

**- FAILURE EFFECTS -****(A) SUBSYSTEM:**LOSS OF ONE OF TWO SERIES RELAYS CAUSING LOSS OF PLBM AC BUS 2 IN MID MOTOR  
CONTROL ASSEMBLY #4. ALSO, FOR SHORT TO GROUND (COIL) ASSOCIATED CIRCUIT  
PROTECTION FUSES TO ONE POLE OF THE PAYLOAD BAY MECHANICAL POWER (SYSTEM  
2) SWITCH WILL OPEN CAUSING LOSS OF PLBM AC BUS 2 IN MID MOTOR CONTROL  
ASSEMBLY #3.

**FAILURE MODES EFFECTS ANALYSIS (FMEA) - NON-CIL FAILURE MODE  
NUMBER: 05-6-2756-01****(B) INTERFACING SUBSYSTEM(S):**

LOSS OF REDUNDANCY. ALL CRITICAL FUNCTIONS HAVE REDUNDANT MOTORS POWERED FROM A DIFFERENT AC BUS IN A DIFFERENT MID MOTOR CONTROL ASSEMBLY AND PLBM AC BUS 2 IN MID MOTOR CONTROL ASSEMBLY #3 DOES NOT POWER MOTORS FOR THE SAME CRITICAL FUNCTIONS.

**(C) MISSION:**

POSSIBLE EARLY MISSION TERMINATION

**(D) CREW, VEHICLE, AND ELEMENT(S):**

FIRST FAILURE - NO EFFECT

**(E) FUNCTIONAL CRITICALITY EFFECTS:**

POSSIBLE LOSS OF CREW/VEHICLE AFTER SECOND FAILURE (LOSS OF REDUNDANT MOTOR OR POWER/CONTROL CIRCUIT) DUE TO THE LOSS OF CAPABILITY TO SAFELY LATCH/RELEASE PAYLOADS.

**DESIGN CRITICALITY (PRIOR TO DOWNGRADE, DESCRIBED IN (F)):****(F) RATIONALE FOR CRITICALITY DOWNGRADE:**

THE DESIGN CRITICALITY OF 1R2 HAS BEEN DOWNGRADED TO 1R3 AFTER WORKAROUNDS CONSIDERATION (ALLOWED PER CR S050107W) BECAUSE AFTER THE SECOND FAILURE, EVA CAN BE PERFORMED TO MANUALLY LATCH/RELEASE THE PAYLOAD LATCHES.

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**- APPROVALS -**

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EDITORIALLY APPROVED  
TECHNICAL APPROVAL

: BNA  
: VIA APPROVAL FORM

: J. Kamura 7-26-99  
: 96-CIL-025\_05-8