

**FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CIL HARDWARE**  
**NUMBER: 05-8BA-2365-IM -X**

**SUBSYSTEM NAME: EPD&C - LANDING GEAR CONTROL**  
**REVISION: 6 03/30/92**

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**PART DATA**

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	<b>PART NAME</b>	<b>PART NUMBER</b>
	<b>VENDOR NAME</b>	<b>VENDOR NUMBER</b>
LRU	: PANEL R4	V070-730278
SRU	: RESISTOR	RWR80S1211FR

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**EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:**  
RESISTOR, LG/NWS HYDRAULIC SYSTEM AUTO 1/2 MDM STATUS (1.2K, 2W)

**REFERENCE DESIGNATORS: 32V73A4A1R1**

**QUANTITY OF LIKE ITEMS: 1**  
ONE

**FUNCTION:**  
PROVIDES CURRENT LIMITING BETWEEN CONTROL BUS BC1 AND MDM, AND ISOLATION ABOVE GROUND BY 1.2K OHMS.

**FAILURE MODES EFFECTS ANALYSIS FMEA - CIL FAILURE MODE**

**NUMBER: 05-6BA-2365-IM-02**

**REVISION#: 7 07/01/99**

**SUBSYSTEM NAME: EPD&C - LANDING GEAR CONTROL**

**LRU: PANEL R4**

**CRITICALITY OF THIS**

**ITEM NAME: RESISTOR**

**FAILURE MODE: 1R3**

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**FAILURE MODE:**

**SHORT (END TO END)**

**MISSION PHASE:**

LO LIFT-OFF  
OO ON-ORBIT  
DO DE-ORBIT

**VEHICLE/PAYLOAD/KIT EFFECTIVITY:**

102 COLUMBIA  
103 DISCOVERY  
104 ATLANTIS  
105 ENDEAVOUR

**CAUSE:**

STRUCTURAL FAILURE (MECHANICAL STRESS, VIBRATION), CONTAMINATION,  
ELECTRICAL STRESS, THERMAL STRESS, PROCESSING ANOMALY

**CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO**

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**REDUNDANCY SCREEN**

A) FAIL  
B) N/A  
C) PASS

**PASS/FAIL RATIONALE:**

A)

FAILS SCREEN "A" BECAUSE RESISTOR SHORT (END TO END) IS NOT CAPABLE OF BEING TESTED DURING GROUND TURNAROUND.

B)

SCREEN "B" IS "N/A" SINCE THIS RESISTOR IS CONTAINED WITHIN A STANDBY SYSTEM.

C)

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**- FAILURE EFFECTS -**

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**(A) SUBSYSTEM:**

LOSS OF ISOLATION BETWEEN CONTROL BUS BC1 AND MDM

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**(B) INTERFACING SUBSYSTEM(S):  
FIRST FAILURE - NO EFFECT**

**(C) MISSION:  
FIRST FAILURE - NO EFFECT**

**(D) CREW, VEHICLE, AND ELEMENT(S):  
FIRST FAILURE - NO EFFECT**

**(E) FUNCTIONAL CRITICALITY EFFECTS:  
SUBSEQUENT FAILURE (LG/NWS SWITCH SHORTS TO GROUND) CAUSES LOSS OF  
CONTROL BUS BC1. LOSS OF EITHER CONTROL BUS BC3 OR CA2 CAUSES LOSS OF THE  
NLG EXTENSION PYRO ASSIST CAPABILITY RESULTING IN POSSIBLE LOSS OF  
CREW/VEHICLE.**

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**-DISPOSITION RATIONALE-**

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**(A) DESIGN:  
REFER TO APPENDIX E, ITEM NO. 3 - RESISTOR, WIRE WOUND**

**(B) TEST:  
REFER TO APPENDIX E, ITEM NO. 3 - RESISTOR, WIRE WOUND**

**GROUND TURNAROUND TEST  
NONE**

**(C) INSPECTION:  
REFER TO APPENDIX E, ITEM NO. 3 - RESISTOR, WIRE WOUND**

**(D) FAILURE HISTORY:  
CURRENT DATA ON TEST FAILURES, FLIGHT FAILURES, UNEXPLAINED ANOMALIES, AND  
OTHER FAILURES EXPERIENCED DURING GROUND PROCESSING ACTIVITY CAN BE  
FOUND IN THE PRACA DATA BASE.**

**(E) OPERATIONAL USE:**

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| CORRECTIVE ACTION IN THE EVENT OF A FAILURE IS NONE

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- APPROVALS -

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EDITORIALLY APPROVED  
TECHNICAL APPROVAL

: BNA  
: VIA APPROVAL FORM

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