CIL EMU CRITICAL ITEMS LIST			5/30/2002 SUPERSEDES 12/31/2001		Page 1 Date: 6/5/2002
NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE	
		106FM04Y			
RESTRAINT, PHASE VI, ITEM 106 (1) LEFT (1) RIGHT 	2/2	106FM04Y Loss of convolute positioning or failed tight line or lower gimbal sheath. Defective material, webbing or thread. Material abrasion.	END ITEM: Ballooning of convolute in areas of non- adhesion restricting wrist mobility or loss of position of soft wrist. GFE INTERFACE: Hampered wrist mobility. Crewman fatigue. MISSION: Terminate EVA. CREW/VEHICLE: None. TIME TO EFFECT /ACTIONS: Minutes. TIME AVAILABLE: N/A TIME REQUIRED: N/A REDUNDANCY SCREENS: A-N/A B-N/A C-N/A	A. Design - The soft wrist design employs the use of fabric folds in the proper location. T 1000 webbing and slide over the Teflon lines are anchored in two places using The lower gimbal sheaths consist of a 6 basted to a 6.0 oz polyester fabric she cord captured in the seam allowance alc assemblies around the circumference of the soft wrist relative to the lower gi B. Test - Component Acceptance Testing: PDA: The following tests are conducted at the ILC Document 0111-710112. 1. Proof pressure test at 8.0 (+ 0.2 damage. Certification: The glove restraint assembly was success certifiation testing to duplicate opera Report for the Phase VI Glove, ILC Doc. reflecting requirements of significance documented during certification testing certification while the actual indicate restraint in the Hamilton Sundstrand Li Requirements S/AD 	<pre>f two "tight" lines that ensure the the tight lines are made of 1/2" Spec covered primary webbings. The tight size "E" polyester thread. .0 oz polyester fabric reinforcement ath. The sheath includes a polyeste ng each side. There are four sheath the lower gimbal. The assemblies 1c mbal in order to provide smooth move e glove assembly level in accordance - 0.0) psig to verify no structural sfully tested (manned) during tional usage (Ref. Certification Tes 0111-712701). The following usage, to the glove restraint assembly, we to the s/AD applies 229 hours in s 198 hours toward the Phase VI glov mited Life Items list (EMU1-19-001). Actual 14830 10830 17393 198 99 63 18 49</pre>
				Components and material manufactured to are documented from procurement through receiving inspection verifies that the the procurement documents, that no dama supplier certifications have been recei information.	ILC requirements at an approved sup shipping by the supplier. ILC incomaterials received are as identified ge has occurred during shipment and ved which provide traceability

The following MIP's performed during the glove assembly manufacturing proce assure that the failure causes are precluded from the fabricated item:

CIL EMU CRITICAL I	TEMS LIST		5/30/200 12/31/20	2 SUPERSEDES 01	Page 2 Date: 6/5/2002
NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE	
		106FM04Y			
				 Verification of visual inspection of convolution holes. Verification of seam acceptability. 	ute for absence of tears and
				During PDA, the following inspection points are level in accordance with ILC Document 0111-7101	performed at the glove asse 12:
				 Visual inspection for material degradation. Visual inspection for structural damage after 	er proof-pressure tests.
				D. Failure History - None.	
				E. Ground Turnaround - During ground turnaround, in accordance with FEI (with TMG removed), bladder (installed) is visu for structural integrity, material damage or deg structural test is also performed.	MU-R-001, the glove assembly ally inspected to extent pos gradation. Glove and EMU le
				F. Operational Use - Crew Response - Pre/Post EVA: If during airlock operations, rej backup gloves. EVA: If hand fatigue, terminate EVA.	press airlock. Consider use
				Special Training - Standard training covers this failure mode.	
				Operational Considerations - Flight rule A15.1.2-2 of "Space Shuttle Operation defines go/no go criteria related to EMU pressus Checklist, JSC-48023, procedures Section 3 (EMU verify hardware integrity and systems operationa Time Data System allows ground monitoring of EMM	onal Flight Rules", NSTS-128 re integrity. Generic EVA Checkout) and 4 (EVA prep) al status prior to EVA. Rea U systems.

EXTRAVEHICULAR MOBILITY UNIT

SYSTEMS SAFETY REVIEW PANEL REVIEW

FOR THE

I-106 GLOVE ASSEMBLY

CRITICAL ITEM LIST (CIL)

EMU CONTRACT NO. NAS 9-97150

Aluman Jr SS - Project Engineering Prepared by:

<u>III. Sniplin</u> HS - Reliability

Approved by: NASA – SSA/SSM 2244

5/23/cr

R. Munford 4/24/02 HS - Engineering Manager

Che & In 6/3/02

6/05/02 6/3/02 um MASA - Crew

rogram Manager