CIL

EMU CRITICAL ITEMS LIST

5/30/2002 SUPERSEDES 12/31/2001

Page 1

Date: 3/27/2002

NAME		FAILURE		
QTY	CRIT	CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
		113AFM03		
CHECK VALVE, ITEM 113A	2/2	External gas leakage, fill line side. Seal failure.	END ITEM: Oxygen leakage to ambient when the 02 fill line is attached and pressurized for recharge and IVA.	A. Design - Leakage path is through radial type silicone O-seal. Seal design configuration,
SV778873-15 (1)				dimensions, and rigidness of assembly provide squeeze under all load conditions.
				B. Test - Component Acceptance Test - The manufacturer, CTI, performs an external leakage test to assure seal integrity.
				PDA Test -
			GFE INTERFACE: Depletion of the vehicle	The oxygen check valve is leak tested by pressurizing the bottles to 850-950 psia with a mixture of 98% N2 and 2% He. A helium mass spectrometer is then used to "sniff" for evidence of external leakage.
			oxygen Suppry.	Certification Test - Certified for a useful life of 20 years (Ref. EMUM-0083).
			MISSION: Loss of use of one EMU.	C. Inspection - Details, including the O-ring, O-ring grooves and sealing surfaces, are 100% inspected per drawing dimensions and surface finish characteristics. Details are
			CREW/VEHICLE: None.	running and final torques of all threaded connectors are verified by vendor and DCAS inspection, a trial assembly is run on all details and then they are visually inspected.
			TIME TO EFFECT	
			Minutes.	None.
			TIME	
			AVAILABLE:	E. Ground Turnaround -
			N/A	Tested for non-EET processing per FEMU-R-UUI, 02 Fill Line and Item 113C External Leakage. None for EET processing.
			TIME REQUIRED:	Internal Ioanager None for LIT proceeding?
			N/A	F. Operational Use -
			REDUNDANCY	Crew Response - PreEVA: No response, failure unlikely to be detectable on orbit.
			SCREENS:	PostEVA: No response, failure unlikely to be detectable on orbit.
			A-N/A	Training - No training specifically covers this failure mode.
			B-N/A C-N/A	Operational Considerations - Flight rules define require EVA termination when minimum primary consumables
				remain.
				EVA checklist procedures verify hardware integrity and systems operational status prior to EVA.

EXTRAVEHICULAR MOBILITY UNIT

SYSTEMS SAFETY REVIEW PANEL REVIEW

FOR THE

I-113 PRIMARY PRESSURE CONTROL MODULE

CRITICAL ITEM LIST (CIL)

EMU CONTRACT NO. NAS 9-97150

Prepared by: AIS - Project Engineering Approved by: APR Approved by: APR Approved by: APR ASA

<u>M. Snyh</u> HS - Reliability

Ula Plough for foru HS - Engineering Manager

6/25/02

<u>Sm 1/27/02</u> <u>Sm 1/27/02</u> <u>- 71./22</u> NASA

NASA - Crew

1 mathen J. Miller 7-102 NASA Frogram Manager