CIL EMU CRITICAL ITEMS LIST			5/30/2002 SUI	PERSEDES 12/31/2001	Page 1 Date: 3/27/2002	
NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE		
		113BFM01	. – – – – – –			
FLOW RESTRICTOR, ITEM 113B	2/1R	No flow, clogs. Contamination. Adjusting screw rotates and orifice slides out of position.	END ITEM: Reduction and/or loss of primary oxygen flow to the regulators.	A. Design - The inlet from the tanks is protected by a 25 micron filter. Inlet for check valve is protected by 25 micron filters. The adjusting screw wired against rotation in two directions and has a KEL-F locking plus		
			GFE INTERFACE: Drop in suit and water reservoir	B. Test - Component Acceptance Test (Vendor) - The manufacturer, CTI, tests the maximum flow at 1035-1040 the flow is equal to or less than 7.5 lb/hr. The regulator stability tests require a 5.2 lb/hr flow.	tests the maximum flow at 1035-1040 psig, to assure that r less than 7.5 lb/hr. The regulator performance and e a 5.2 lb/hr flow. test of the flow restrictor. With the bottles pressurized ifice is required to maintain a maximum flow of 5.5-6.78 A clogged orifice would fail this test. from becoming contaminated, all rig lines, gases, and	
			pressure. The SOP will automatically deliver emergency oxygen when the suit drops	PDA Test - SEMU-60-010 contains a test of the flow restrictor. With th to 850-950 psia the orifice is required to maintain a maxim maximum lbs/hr oxygen. A clogged orifice would fail this te To prevent the orifice from becoming contaminated, all rig test fixtures are cleaned to HS3150 EM50A.		
			below 3.33 psia minimum during EVA.	Certification Test - Certified for a useful life of 20 years (Ref. EMUM-0083). C. Inspection -	fied by the worder	
			MISSION: Terminate EVA. Loss of use of one EMU.	The running and final torque of the adjusting screw is veri and DCAS inspection. A trial assembly is performed and then visually inspected. The safety wire is inspected after asse	n the details are	
				D. Failure History - None.		
			CREW/VEHICLE: None for			
			single failure. Possible loss of crewman with loss of	Ground Turnaround - sted for non-EET processing per FEMU-R-001, IV1103 Performance Data ar Regulator Check. FEMU-R-001 Para 8.2 EMU Preflight KSC Checkout for occssing.		
			SOP.	F. Operational Use - Crew Response - ProFVA: Trouble cheet problem if no success consider FMIL 3) if arailable EMU no	
			TIME TO EFFECT /ACTIONS: Immediate.	PreEVA: Trouble shoot problem, if no success consider EMU 3 go for EVA. EVA: When CWS data confirms loss of suit and feedwater presterminate EVA. Training - Standard EMU training covers this failure mode.		
			TIME AVAILABLE: Minutes.	Operational Considerations - Flight rules define go/no go criteria related to EMU suit p Flight rules require termination of EVA if SOP activated. EVA checklist procedures verify hardware integrity and oper	-	
			TIME REQUIRED:	to EVA. Real Time Data System allows ground monitoring of E		

REDUNDANCY SCREENS:

TIME REQUIRED: Immediate

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A-PASS B-PASS C-PASS

EXTRAVEHICULAR MOBILITY UNIT

SYSTEMS SAFETY REVIEW PANEL REVIEW

FOR THE

I-113 PRIMARY PRESSURE CONTROL MODULE

CRITICAL ITEM LIST (CIL)

EMU CONTRACT NO. NAS 9-97150

Prepared by: Approved by: RMS - Project Engineering Approved by: RMSA - SSM