CIL

EMU CRITICAL ITEMS LIST

## 5/30/2002 SUPERSEDES 12/31/2001

Page 1

Date: 3/27/2002

----

NAME		FAILURE		
P/N		MODE &		
QTY	CRIT	CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
		113EFM05		
WATED DDECCIIDE	2/10	Evternal das	END TTEM.	A Design -
REGULATOR, ITEM 113E	2/1K	leakage.	Leakage of primary oxygen supply to	Leak is through a radial O-seal on the bellows cap, two radial O-seals at the 15 psi outlet, and the bellows. The seal design configuration, dimensions and rigidness of assembly provide squeeze under all load conditions. The bellows has
SV778873-15 (1)		Seal failure, bellows	ambient.	a proof pressure of 84 psi.
		leakage.	GFE INTERFACE:	B. Test -
			Excessive	Vendor Component Acceptance Test -
			consumption of the primary	The regulator manufacturer, Carleton performs an external leakage test to assure bellows and seal integrity.
			The SOP is	PDA Test -
			automatically	An external leakage test per SEMU-6-010 verifies bellows and seal integrity.
			activated during EVA if	With the regulator outlet pressurized to 14.6 - 15.7 psig using 98% N2 and 2% He, a helium sniff test must show no evidence of leakage.
			the suit	
			pressure drops	Certification Test -
			minimum.	Cercified for a useful file of 20 years (Ref. EMOM-0083).
				C. Inspection -
			MISSION:	Details are 100% inspected per drawing dimensions and surface finish
			Terminate EVA.	characteristics. Details are manufactured from material with certified physical
			Loss of use of one EMU.	and chemical properties. All details, gases and test facilities are cleaned and inspected to HS3150 EM50A to preclude contamination clogging. The running and final torgues of all threaded connections are verified by Vendor and DCAS
			CREW/VEHICLE:	inspection. A trial assembly is run on all details and then they are visually isspected
			single	
			Possible loss	D. Failure History -
			of crewman	None.
			with loss of	
			SUF.	E. Ground Turnaround -
			TIME TO EFFECT /ACTIONS:	Tested for non-EET processing per FEMU-R-001, Final SEMU Gas Structural and Leakage. None for EET processing.
			Immediate.	E (marational line
			TME	r. Operational Use -
			AVAILABLE:	PreEVA: When detected during suit leak check, trouble shoot problem, if no
			Minutes.	success consider EMU 3 if available. EMU no go for EVA. EVA: When CWS data confirms an accelerated primary 02 use rate, terminate EVA.
			TIME REQUIRED:	If CWS data confirms a loss of suit pressure integrity coupled with an
			Immediate.	accelerated primary 02 use rate, abort EVA. Training – Standard EMU training covers this failure mode.
			REDUNDANCY	Operational Considerations -
			SCREENS:	Flight rules define go/no go criteria related to EMU suit pressure regulation.
			A-PASS B-DASS	Consider periodic vacuum U2 recharge to recover EMU operation.
			D-FASS C-PASS	status prior to EVA. Real Time Data System allows ground monitoring of EMU
			2 1100	systems.

## EXTRAVEHICULAR MOBILITY UNIT

## SYSTEMS SAFETY REVIEW PANEL REVIEW

FOR THE

I-113 PRIMARY PRESSURE CONTROL MODULE

CRITICAL ITEM LIST (CIL)

## EMU CONTRACT NO. NAS 9-97150

Prepared by: AIS - Project Engineering Approved by: APR Approved by: APR Approved by: APR ASA

<u>M. Snyh</u> HS - Reliability

Ula Plough for foru HS - Engineering Manager

6/25/02

<u>Sm 1/27/02</u> <u>Sm 1/27/02</u> <u>- 71./22</u> NASA

NASA - Crew

1 mathen J. Miller 7-102 NASA Frogram Manager