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CIL

EMU CRITICAL ITEMS LIST 5/30/2002 SUPERSEDES 12/24/1992 Date: 3/27/2002

			0,00,2002	Date: 3/27/2002	
NAME		FAILURE			
P/N QTY	CRIT	MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE	
		115FM07			
SHEAR PLATE ASSEMBLY, ITEM 115 (PIVOTED, PLANAR) SV778540-56 (1) OR (ORU) SV824133-8 (1)	2/2	T15FM07 Fails to open PLSS shutoff valve.  Linkage mechanism deformed.	END ITEM: Failure to open primary oxygen regulator shutoff valve when 02 actuator is not in OFF position. Blockage of 02 flow path from oxygen tank to primary regulator.  GFE INTERFACE: Unable to pressurize the EMU with PLSS.  MISSION: Loss of use of	A. Design - The Item 115 shear plate contains the actuator cam which opens the PLSS shutor valve poppet using the cam rise and direct linkages. The actuator link to the shutoff has a load capacity to yield stress of 160 lb while the actuation maximum force of the shutoff valve is 8 lb.  B. Test - Component Acceptance Test - None.  PDA Test - None.  Certification Test - Certified for a useful life of 20 years from the date of manufacture. Successful refurbishment will extend useful life to 30 years max. (ref EMUM1-0491, EMUM1-0027).  C. Inspection - Details are 100% inspected per drawing dimensions and surface finish characteristics. Details are manufactured from material with certified physicand chemical properties. All details, gases and test facilites are cleaned as inspected to HS3150 EM50A to preclude contamination clogging.  D. Failure History -	cal
			one EMU. CWS issues a low suit pressure	None.	
			warning.	E. Ground Turnaround - Tested for non-EET processing per FEMU-R-001, V1103 Performance Data and Item 113 Regulator Check. FEMU-R-001 Para 8.2 EMU Preflight KSC Checkout for EET	
			CREW/VEHICLE: None.	processing.  F. Operational Use - Crew Response -	
			TIME TO EFFECT /ACTIONS: None.	Pre/PostEVA: Trouble shoot problem. If no success, terminate EVA and consider use of third EMU if available. EMU is no go for EVA.  Training - Standard training covers this failure mode.  Operational Considerations - Flight rules define loss of EMU for loss of pressure regulation. EVA checklis	
			TIME AVAILABLE: N/A	procedures verify hardware integrity and systems operational status prior to 1	
			TIME REQUIRED: N/A		
			REDUNDANCY		

SCREENS: A-N/A B-N/A C-N/A

## EXTRAVEHICULAR MOBILITY UNIT

## SYSTEMS SAFETY REVIEW PANEL REVIEW

FOR THE

I-115 SHEAR PLATE ASSEMBLY

CRITICAL ITEM LIST (CIL)

EMU CONTRACT NO. NAS 9-97150

Prepared by: Approved by: Approved by: Approved by: