CIL

EMU CRITICAL ITEMS LIST

5/30/2002 SUPERSEDES 12/24/1992

Page 1

Date: 3/27/2002

AME		FAILURE MODE &		
ТҮ	CRIT	CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
		115FM08		
HEAR PLATE SSEMBLY, ITEM 15 (PIVOTED, LANAR) V778540-56 1)	2/2	Fails to close PLSS 02 shutoff valve. Linkage malfunction.	END ITEM: Failure to close primary oxygen regulator shutoff valve when 02	A. Design - The shutoff valve linkage consists of a spring loaded translating push rod assembly which is pinned to a lever. Output motion of the lever moves the spring loaded shutoff valve. The linkage springs provide a force of 17.6 lbs move the linkage towards the shutoff position. In addition, the shutoff valve itself is forced closed by a 5 lb spring force for a total available force of 22.6 lbs to move the linkage toward the shutoff position.
R (ORU)			actuator is in off position.	Linkage system friction is kept low by using a ball bearing guide at the cam e of the push rod and clevis pin joints with good wheelbase on the lever.
SV824133-8 (1)			GFE INTERFACE: Primary oxygen supply continues to flow to	Thus a net force moving over 20 lbs exists to prevent linkage hangup in the shutoff valve open position. B. Test -
			primary suit regulator when 02 actuator is off. PLSS oxygen is	Component Acceptance Test - Shear plate level testing per AT-E-115 verifies the shutoff valve closed by para. 9.0 leakage test in which 1.67 x 10 -5 scc/sec He maximum. It is allowe at 900 psi; and verifies proper actuator operation by para. 10 in which the force required to slide the actuator to any off positions is 15 lbs max.
			vented into airlock.	PDA Test - Proper actuator mechanism operation is verified during SEMU-60-010 at the PLSS
			MISSION: Loss of use of one EMU. Loss of PLSS bottle pressure during EMU	level identical to the above component level test. Certification Test - Certified for a useful life of 20 years from the date of manufacture. Successful refurbishment will extend useful life to 30 years max. (ref EMUM1- 0491, EMUM1-0027).
			doffing and PLSS recharging.	After which proper PLSS shutoff valve actuation was verified.
			CREW/VEHICLE: None.	C. Inspection - Details are 100% inspected per drawing dimensions and surface finish characteristics. Details are manufactured from material with certified physic and chemical properties.
			TIME TO EFFECT /ACTIONS: None.	D. Failure History - None.
			TIME AVAILABLE:	E. Ground Turnaround - Tested for non-EET processing per FEMU-R-001, High Pressure O2 Leakage. FEMU-F 001 Para 8.2 EMU Preflight KSC Checkout for EET processing.
			N/A TIME REQUIRED:	F. Operational Use - Crew Response - POST EVA: Respress A/L. With SOP deactivated, disable airlock 02 supply to
			N/A REDUNDANCY SCREENS:	SCU. Doff EMU normally allowing primary 02 tanks to vent into cabin. For subsequent EVA's activate airlock 02 supply to SCU prior to fan activation during EMU donning. EMU go for SCU standby or EVA depending on feedwater remaining.
			A-N/A B-N/A	Training -

CIL EMU CRITICAL ITEMS LIST			5/30/2002 SU	PERSEDES 12/24/1992	Page 2 Date: 3/27/2002
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		115FM08			
			C-N/A	No training specifically covers this failure mode.	
				Operational Considerations -	EVA chocklict

Flight rules define loss of EMU for loss of pressure regulation. EVA checklist procedures verify hardware integrity and systems operational status prior to EVA.

EXTRAVEHICULAR MOBILITY UNIT

SYSTEMS SAFETY REVIEW PANEL REVIEW

FOR THE

I-115 SHEAR PLATE ASSEMBLY

CRITICAL ITEM LIST (CIL)

EMU CONTRACT NO. NAS 9-97150

Prepared by: Approved by: RAB L Approved by: RAB L RAB L

Ula Ploye HS - Engineering Manag tor RMa

M. Smych HS - Reliability

. u/mods

Manages