CIL EMU CRITICAL ITEMS	LIST		5/30/2002 SUPERSEDES 12/24/1995		Page 1 Date: 3/27/2002
NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE	
		128FM01			
CHECK VALVE ASSEMBLY, ITEM 128 SV767699-1 (1)	2/2	Fails to close, internal leakage.	END ITEM: Reverse flow path through valve in check direction.	A. Design - The valve has a soft elastameric (silicone) flapper and closed in the check direction. Nominal reverse pressure is provided to 20 microns by the Item 141 Gas Trap Scree	is pressure loaded is 1.0 psi. Filtration n.
		Surface imperfections on flapper or seat, contamination.	GFE INTERFACE: Unable to purge a large gas bubble in the coolant pump,	B. Test - Component Acceptance Test - An internal leakage test is performed per AT-E-127/128 i: outlet is pressurized to 0.9 - 1.1 psid with water. Leak scc/min. PDA Test -	n which the check valve age must not exceed 1.0
			resulting in inability to prime pump. Loss of coolant flow during preEVA EMU Checkout.	A leakage test is run per SEMU-60-010 where the check va 0.9 - 1.1 psid. Leakage must not exceed 3.8 scc/min. H20 check valve is also verified during the coolant loop dry with an empty cooling circuit, the PLSS water pump must 1 the cooling circuit and remove all gas bubbles within 10 pump. A failed open check valve would prevent the pump f	lve is pressurized to . Proper function of the charge test. Starting be capable of filling minutes of starting the rom operating properly.
				Certification Test - Certified for a useful life of 25 years (ref. EMUM1-0023).
			MISSION: Loss of use of	C. Inspection -	
			one EMU.	Cause - Surface imperfections on flapper or seat. The ch flapper sealing interfaces are 100% inspected to meet di finish requirements.	eck valve housing and mensional and surface
			CREW/VEHICLE:		
			None. Crew discomfort. (hot)	Cause - Contamination. A cleanliness level of HS3150 EM150 is maintained during the check valve. This cleanliness level requires a manda verification.	assembly and testing of tory inspection for
			TIME TO EFFECT /ACTIONS:	D. Failure History - EMU-128-D002 (8-27-80) Internal leakage due to tool mark	s in Kel-F housing.
			Minutes.	Corrective action was to change sealing surface finish r H-EMU-128-D003 (12-12-83) Excessive internal leakage in	equirement. reverse flow direction
			TIME AVAILABLE: N/A	due to surface irregularities on housing face. Corrective component acceptance test to detect small leakage.	e action was to revise
			TIME REQUIRED: N/A	J-EMU-128-A001 (06/02/95) Item 128 Check Valve S/N 004 for test. Spec is 56.5 cc; actual 240 cc per 15 minutes. A (T6 to T4) created a flow path across check valve (in re- invalidating the 128 reverse flow test. Revised the test	ailed reverse leakage ddition of LCVG bypass verse direction) t procedure.
			REDUNDANCY SCREENS: A-N/A B-N/A	E. Ground Turnaround - Tested for non-EET processing per FEMU-R-001, Dry LCVG D processing.	egas Test. None for EET
			C-N/A	F. Operational Use - Crew Response - PreEVA: Trouble shoot problem. If no success, consider t otherwise continue with EVA prep.	hird EMU if available,

CIL EMU CRITICAL ITEMS LIST			5/30/2002 SUE	PERSEDES 12/24/1995	Page 2 Date: 3/27/2002
NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE	
		128FM01			
				Training - Standard training covers this failure mode. Operational Considerations - RTDS allows ground monitoring of EMU systems. EVA check list procedures verify hardware integrity and sys status prior to EVA. Flight rules define loss of EVA for lo	tems operational ss of thermal control.

EXTRAVEHICULAR MOBILITY UNIT

SYSTEMS SAFETY REVIEW PANEL REVIEW

FOR THE

I-128 CHECK VALVE AND HOUSING

CRITICAL ITEM LIST (CIL)

EMU CONTRACT NO. NAS 9-97150

Prepared by: SS - Project Engineering Approved by: MB MB

Min

HS - Engineering Manager

galor

NASA Crew

rogram Manager