

CIL  
CRITICAL ITEMS LIST  
FILE: CIL7/1

NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
ELECTRICAL SIGNALS HARNES ITEM 162 SV709154-2 111	E/IRB	152FM01; ELECTRICAL SHORT: +5V, +10V, -14.2V OR -14.2V LINE TO GROUND.	END ITEM; SHORT FROM +5V, +10V, -14.2V OR +14.2V LINE TO GROUND. LOSS OF POWER.	A. DESIGN - EACH CONNECTOR/CABLE INTERFACE IS STRAIN RELIEVED BY POSITIONING THE CONDUCTOR IN PLACE. A RUBBER BACKSHELL IS THEN MOLDED OVER THE CONNECTOR/CABLE INTERFACE. EACH CONNECTOR/ADAPTER WING INTERFACE IS LOCKED IN PLACE TO PREVENT ROTATION BY A MECHANICAL LOCK AND AN ADHESIVE LOCK. B74 AND TEFLON COATED WIRE PROVIDES THE REQUIRED INSULATION RESISTANCE. THE CONDUCTORS ARE NAWLED WITHIN A NOVEN COPPER STRANDED SHEATH WHICH CAUSES THEM TO ACT TOGETHER AND SHARE ANY LOADING PLACED ON THEM. A NOVEN ADHESIVE SHEATH IS ASSEMBLED OVER THE SHIELD CABLE SHEATHS TO PROVIDE PROTECTION FROM ABRASION AND IMPACT.
102D-1 44		CAUSE: CABLE CHAFING AGAINST CONNECTOR SHELL OR SHIELD. IMPROPER CONNECTOR SRAIN RELIEF.	USE INTERFACE; INCREASE IN BATTERY POWER CONSUMPTION. THE CURRENT IS LIMITED IN THE DCM DC/DC CONVERTER TO 1.0 +/- 0.25 AMPS. SHUTDOWN OF THE DC/DC CONVERTER. LOSS OF CHS, BONES AND DCH DISPLAY.	B. TEST - COMPONENT ACCEPTANCE TEST - FIT 152 HARNES IS SUBJECTED TO ACCEPTANCE TESTING PER AD-611-152 PRIOR TO FINAL ACCEPTANCE TESTING. THIS TESTING INCLUDES THE FOLLOWING TESTS WHICH ENSURE THERE ARE NO WORKMANSHIP PROBLEMS WHICH WOULD CAUSE AN ELECTRICAL SHORT IN THE +5V, +10V AND +14.2V LINES. THE INSULATION RESISTANCE AND DIELECTRIC STRENGTH BETWEEN EACH CONDUCTOR AND THE SHIELD GROUND IS MEASURED TO ENSURE THERE ARE NO SHORTS. EACH CONNECTOR/CABLE INTERFACE IS FULL TESTED (10 POUNDS) TO DETECT ANY WORKMANSHIP PROBLEMS WHICH COULD CAUSE A SHORT CIRCUIT.
			MISSION: NONE FOR SINGLE FAILURE. TERMINATE EVA WITH LOSS OF DCH DISPLAY, CHS AND ABILITY TO MONITOR OPERATIONAL INTEGRITY OF EMI. LOSS OF USE OF ONE EMI.	PDA TEST - THE +5V, +10V AND +14.2V LINES ARE FUNCTIONALLY CHECKED DURING PLS3 PDA TESTING PER SEMU-40-010, PARA 25.6, TO INSURE THERE ARE NO SHORTS TO SHIELD GROUND WHICH AFFECT THE PERFORMANCE OF THE PLS3.
			CREW/VEHICLE: NONE FOR SINGLE FAILURE. POSSIBLE LOSS OF CREWMAN WITH LOSS OF CEC, OXYGEN OR LOW VENT FLOW.	CERTIFICATION TEST - THIS ITEM HAS COMPLETED THE STRUCTURAL VIBRATION AND SHOCK CERTIFICATION REQUIREMENTS DURING 10/83. ENGINEERING CHANGE 47004-527-2 (ADDED A CONNECTOR FULL TEST) HAS BEEN INCORPORATED AND CERTIFIED SINCE THIS CONFIGURATION HAS CERTIFIED.

CIG  
CRITICAL ITEMS LIST  
FILE: 03L7/3

NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
ELECTRICAL SIGNALS HARNESS ITEM 852 SV789152-2 111	2/LAB	152FM031 ELECTRICAL SHORT, 15V, +18V, -14.2V ON +14.2V LINE TO GROUND.		<p>C. INSPECTION - TO ENSURE THERE ARE NO WORKMANSHIP PROBLEMS WHICH WOULD CAUSE A SHORT CIRCUIT ON THE HARNESS CONDUCTORS, THE FOLLOWING INSPECTIONS ARE PERFORMED: HARNESS CABLES AND CONDUCTORS ARE VISUALLY INSPECTED PRIOR TO ASSEMBLY TO ENSURE THERE ARE NO DEFECTS WHICH WOULD CAUSE A SHORT TO GROUND DUE TO DEFECTS IN THE CABLE INSULATION. CONNECTOR WIRING IS INSPECTED BEFORE AND AFTER POTTING TO ENSURE THERE IS NO CONDUCTOR BRIDGE AND THAT THE CONDUCTORS ARE PROPERLY STRAIN RELIEVED AND PROPERLY DRESSED TO PREVENT CONDUCTOR SHORTING TO THE ADJACENT WIRE. INSULATION RESISTANCE AND DIELECTRIC STRENGTH ARE MEASURED BETWEEN EACH CONDUCTOR AND SHIELD GROUND TO ENSURE THERE ARE NO SHORTS PRIOR TO AND AFTER POTTING OF THE CONNECTORS. IN-PROCESS ELECTRICAL CHECKOUT OF THE HARNESS BEFORE AND AFTER POTTING AND WELDING TO ENSURE THERE ARE NO SHORT CIRCUITS.</p> <p>D. FAILURE HISTORY - H-EMU-152-001 (2/9/84) DURING PLS ACCEPTANCE TESTING, ALL SENSOR OUTPUTS READ FULL SCALE. A SHORT CIRCUIT IN THE HARNESS WAS FOUND BETWEEN WREF AND GROUND. THE SHORT WAS DUE TO IMPROPER ASSEMBLY AND TESTING BY THE VENDOR. THE VENDOR'S ASSEMBLY AND TEST PROCEDURES WERE REVISED. RELATED FAILURE: J-EMU-152-002 (4-11-85) - DURING A PRE-FLIGHT COMMUNICATIONS CHECK, IT WAS NOT POSSIBLE TO TRANSMIT THROUGH THE RIGHT MICROPHONE ON THE CCA. THE FAILURE WAS CAUSED BY A SHORT CIRCUIT BETWEEN THE RIGHT MICROPHONE PINGER LINE AND THE CABLE GROUNDING SHIELD. THE INSULATION ON THE PINGER LINE HAS BEEN DAMAGED PRIOR TO THE CABLE ASSEMBLY. CC 42004-927-2 WAS ISSUED TO CORRECT THE SV789152-2 HARNESS CONFIGURATION BY ADDING A CONNECTOR PULL TEST TO THE ACCEPTANCE TESTING REQUIREMENTS. FIELD UNITS WERE TESTED TO THE SAME PROCEDURES PULL AND IR MEASUREMENTS PER SI-EMU-300.</p> <p>E. GROUPED BURST/SHOCK - GROUND TURNOVER TESTED PER JEMU-R-001, BCM DISPLAY.</p>
1B24-2 **				

CFL  
CRITICAL ITEMS LIST  
FILE: CIL7/A

NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
ELECTRICAL SIGNALS HARNES ITEM 152 SV709152-2 11	R/IRB	ISSUES: ELECTRICAL SHORT, 15V, +15V, -14.2V OR +14.2V LINE TO GROUND.		<p>P. OPERATION USE CREW RESPONSE PRE-EVA : TROUBLE SHOOT, IF NO SUCCESS CONSIDER THIRD CREW IF AVAILABLE. OTHERWISE, EMU IS NO-GO FOR EVA. EVA : TERMINATE EVA WHEN DETECTED BY GROUND OR DURING CREWMEMBER'S STATUS CHECK. POST-EVA: N/A TRAINING STANDARD EIU TRAINING COVERS THIS FAILURE MODE. OPERATIONAL CONSIDERATIONS REFERENCE LOSS/FAILURE FLIGHT RULES: DEFINE EIU AS LOST IF CREW AND GROUND DETERMINE INSUFFICIENT CDS DATA AVAILABLE. EVA CHECKLIST AND FDP PROCEDURES VERIFY HARDWARE INTEGRITY AND SYSTEMS OPERATIONAL STATUS PRIOR TO EVA. REAL TIME DATA SYSTEM ALLOWS GROUND MONITORING OF EIU SYSTEMS.</p>

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