CIL

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EMU CRITICAL ITEMS LIST			5/30/2002 St	JPERSEDES 12/31/2001	Date: 3/27/2002	
NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE		
		170FM01				
VENT TUBE ASSEMBLY, ITEM	MBLY, ITEM 170	External leak.	END ITEM: Suit gas	A. Design - The O-seals at inlet and outlet interface connects		
SV809669-3 (1)		Seal Failure.	leakage to ambient.	misalignment. The seals are radial type O-rings made from elastomeric mater (viton). O-ring seal configuration and rigidness of assembly provide squeez under all loading conditions. Corrosion protection is provided by a vespel adapter tube fitted between the inconel ventilation tube and the aluminum		
			GFE INTERFACE: Excessive	fan/pump/separator housing. The valve module inte corrosion resistant stainless steel against the in	erface at the other end is	
SV809670-1 (1)			consumption of the primary	B. Test -		
			oxygen supply. The SOP is automatically	Component Acceptance Test - The tube must withstand $10.0 +/- 0.5$ psig nitroger no visable leakage while immersed in Freon.	n for two minutes minimum with	
			activated during EVA if the suit pressure drops	PDA Test - An external leakage test is performed per SEMU-60- pressurized to 18.9 - 19.1 psia with oxygen. Measu		
			to 3.33 psia.	4.66 scc/minute.	area reakage is not to exceed	
			MISSION: Terminate EVA. Loss of use of	Certification Test - Certified for a useful life of 15 years (ref. SEM	U-46-004).	
			one EMU.	C. Inspection - The interfacing surfaces between the ventilation t	tube, adapter tube, fan outlet,	
			CREW/VEHICLE: None for a single failure,	and the valve module inlet are 100% inspected to meet dimensional and sur finish requirements. The O-seals are 100% inspected for surface characte per SVHS3432.		
			possible loss of crewman	D. Failure History - (P/N SV785890-Muffler Obsolete Configuration):		
			with loss of SOP. (Second failure)	H-EMU-170001 (6-1-87) Material in "O" seal groove area loss due EC163402-90 Incorporates application of BR127 to Elbow O-Seal groomeduce galvanic corrosion potential on new and field mufflers.	Elbow O-Seal groove area to	
			TIME TO EFFECT /ACTIONS: Seconds. If	(P/N SV809669/SV809670, Vent Tube Assembly): None.		
			EVA, return to the vehicle.	E. Ground Turnaround - Tested for non-EET processing per FEMU-R-001, Final Leakage. None for EET processing.	al SEMU Gas Structural and	
			TIME			
			AVAILABLE: Minutes.	F. Operational Use - Crew Response - PreEVA: Trouble-shoot problem, if no success cons	sider EMU 3 if available.	
			TIME REQUIRED: Immediate.	EMU no go for EVA EVA: When CWS data confirms an accelerated primar If CWS data confirms an accelerated primary 02 use		
			REDUNDANCY	suit pressure regulation, Abort EVA.		

SCREENS:

Training - Standard EMU training covers this failure mode. A-PASS

B-PASS

Operational Considerations -C-PASS

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170FM01

Flight rules define go/no go criteria related to EMU suit pressure regulation. Flight rules require termination of EVA upon activation of SOP. EVA checklist and FDF procedures verify hardare integrity and systems operational status prior to EVA. Real Time Data System allows ground monitoring of EMU systems.

EXTRAVEHICULAR MOBILITY UNIT SYSTEMS SAFETY REVIEW PANEL REVIEW

FOR THE

I-170 VENT TUBE ASSEMBLY

CRITICAL ITEM LIST (CIL)

EMU CONTRACT NO. NAS 9-97150

Prepared by: Approved by: MB - 5hke
NASA - 85A/SSM

M. Style
HS - Reliability

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