CRITICAL ITEMS LIST

PROJECT: SRMS
ASS'Y NOMENCLATURE: SERVO POWER AMPLIFIER
SYSTEM: ELECTRICAL SUBSYSTEM
ASS'Y P/N: 5114DF1177
SHEET: 1

ITEM REF.	REV.	NAME, QTY. & DRAWING REF. DESIGNATION	FAILURE MODE AND CAUSE	FAILURE EFFECT ON END ITEM	HOUR / FUNC. 1/1 CRITICALITY	RATIONALE FOR ACCEPTANCE
2560	1	MOTOR CONTROL LOOP QTY-6 SCHEMATIC 2563721 AND 2563722	MODE: OUTPUT FROM ANALOG DRIVE SIGNAL TO MDA LOWER THAN COMMANDED. CAUSE(S): (1) INPUT SHIFT REGISTER AND LATCH. (2) DIGITAL COMPARATER AND LIMITER. (3) D TO A CONVERTER. (4) ANALOG SUMMER.	ERROR SIGNAL TO MDA IS LIMITED JOINT WILL NOT DRIVE OR WILL DRIVE SLOWER. ARM MOTION MAY RESULT IN AN UNEXPECTED TRAJECTORY. WORST CASE ---------- UNEXPECTED MOTION. SLUGGISH JOINT. UNANNUNCIATED. CREW ACTION REQUIRED. REDUNDANT PATHS REMAINING ---------- N/A		DESIGN FEATURES ---------- THE DESIGN UTILIZES PROVEN CIRCUIT TECHNIQUES AND IS IMPLEMENTED USING CMOS LOGIC DEVICES. CMOS DEVICES OPERATE AT LOW POWER AND HENCE DO NOT EXPERIENCE SIGNIFICANT OPERATING STRESSES. THE TECHNOLOGY IS MATURE, AND DEVICE RELIABILITY HISTORY IS WELL DOCUMENTED. ALL STRESSES ARE ADDITIONALLY REDUCED BY DERATING THE APPROPRIATE PARAMETERS IN ACCORDANCE WITH SPAR-RMS-PA.003. SPECIAL HANDLING PRECAUTIONS ARE USED AT ALL STAGES OF MANUFACTURE TO PRECLUDE DAMAGE/STRESS DUE TO ELECTROSTATIC DISCHARGE. ALL RESISTORS AND CAPACITORS USED IN THE DESIGN ARE SELECTED FROM ESTABLISHED RELIABILITY (ER) TYPES. LIFE EXPECTANCY IS INCREASED BY ENSURING THAT ALL ALLOWABLE STRESS LEVELS ARE DERATED IN ACCORDANCE WITH SPAR-RMS-PA.003. ALL CERAMIC AND ELECTROLYTIC CAPACITORS ARE ROUTINELY SUBJECTED TO RADIOGRAPHIC INSPECTION. DISCRETE SEMICONDUCTOR DEVICES SPECIFIED TO AT LEAST THE TX LEVEL OF MIL-S-19500. ALL DEVICES ARE SUBJECTED TO RE-SCREENING BY AN INDEPENDANT TEST HOUSE. SAMPLES OF ALL PROCURED LOTS/DATE CODES ARE SUBJECTED TO DESTRUCTIVE PHYSICAL ANALYSIS (DPA) TO VERIFY THE INTEGRITY OF THE MANUFACTURING PROCESSES. DEVICE STRESS LEVELS ARE, DERATED IN ACCORDANCE WITH SPAR-RMS-PA.003 AND VERIFIED BY DESIGN REVIEW. COMPARATORS AND OPERATIONAL AMPLIFIERS ARE STANDARD LINEAR INTEGRATED CIRCUITS WITH MATURE MANUFACTURING TECHNOLOGY. APPLICATION CONSTRAINTS ARE IN ACCORDANCE WITH SPAR-RMS-PA.003. D TO A CONVERTER IS SCREENED TO THE REQUIREMENTS OF SPAR-RMS-PA.003.

PREPARED BY: MEUG
SUPERCEDING DATE: 01 OCT 86
APPROVED:
DATE:

CRITICAL ITEMS LIST

PROJECT: SRMS SYSTEM: ELECTRICAL SUBSYSTEM
ASS'Y NOMENCLATURE: SERVO POWER AMPLIFIER ASS'Y P/N: S114OF1177 SHEET: 2

FMEA REF.	REV.	NAME, QTY, & DRAWING REF. DESIGNATION	FAILURE MODE AND CAUSE	FAILURE EFFECT ON END ITEM	HDWR / FUNC. 1/1 CRITICALITY	RATIONALE FOR ACCEPTANCE
2560	1	MOTOR CONTROL LOOP QTY-6 SCHEMATIC 2563721 AND 2563722	MODE: OUTPUT FROM ANALOG DRIVE SIGNAL TO MDA LOWER THAN COMMANDED. CAUSE(S): (1) INPUT SHIFT REGISTER AND LATCH. (2) DIGITAL COMPARATOR AND LIMITER. (3) D TO A CONVERTER. (4) ANALOG SUMMER.	ERROR SIGNAL TO MDA IS LIMITED JOINT WILL NOT DRIVE OR WILL DRIVE SLOWER. ARM MOTION MAY RESULT IN AN UNEXPECTED TRAJECTORY. WORST CASE ---------- UNEXPECTED MOTION. SLUGGISH JOINT. UNANNUNCIATED. CREW ACTION REQUIRED. REDUNDANT PATHS REMAINING ---------- N/A		ACCEPTANCE TESTS ---------- THE SPA IS SUBJECTED TO THE FOLLOWING ENVIRONMENTAL TESTING AS AN SRU. O VIBRATION: LEVEL AND DURATION - REFERENCE TABLE 4 O THERMAL: PLUS 70 DEGREES C TO -25 DEGREES C DURATION - 1 1/2 CYCLES THE SPA IS THEN TESTED AS PART OF THE JOINTS ACCEPTANCE TESTS (VIBRATION AND THERMAL VACUUM TEST). THE SPA'S/JOINTS UNDERGO RMS SYSTEM TESTS (TP518 RMS STRONGBACK AND TP552 FLAT FLOOR TESTS) WHICH VERIFIES THE ABSENCE OF THE FAILURE MODE. QUALIFICATION TESTS ---------- THE SPA IS SUBJECTED TO THE FOLLOWING SRU QUALIFICATION TEST ENVIRONMENTS. THE SPA WAS ALSO TESTED AS PART OF THE JOINT QUALIFICATION TESTS. O VIBRATION: LEVEL AND DURATION - REFERENCE TABLE 4 O SHOCK: 20G/11 MS/3 AXES (6 DIRECTIONS) O THERMAL VAC: +81 DEGREES C TO -36 DEGREES C (6 CYCLES) 1X10**6 TORR O HUMIDITY: TESTED WITH THE SHOULDER JOINT O EMC: MIL-STD-461 AS MODIFIED BY SL-E-0002 (TEST CE01, CE03, CS01, CS02, CS06, RE01, RE02 (H/B), RS01) FLIGHT CHECKOUT ---------- PDRS OPS CHECKLIST (ALL VEHICLES) JSC 16987

PREPARED BY: MEWG SUPERCEDING DATE: 01 OCT 86 APPROVED BY: DATE:

CRITICAL ITEMS LIST

PROJECT: SRMS
ASS'Y NOMENCLATURE: SERVO POWER AMPLIFIER
SYSTEM: ELECTRICAL SUBSYSTEM
ASS'Y P/N: 51T40FT177
SHEET: 3

FMEA REF.	REV.	NAME, QTY, & DRAWING REF. DESIGNATION	FAILURE MODE AND CAUSE	FAILURE EFFECT ON END ITEM	HDWR / FUNC. 1/1 CRITICALITY	RATIONALE FOR ACCEPTANCE
2560	1	MOTOR CONTROL LOOP QTY-6 SCHEMATIC 2563721 AND 2563722	MODE: OUTPUT FROM ANALOG DRIVE SIGNAL TO MDA LOWER THAN COMMANDED. CAUSE(S): (1) INPUT SHIFT REGISTER AND LATCH. (2) DIGITAL COMPARATER AND LIMITER. (3) D TO A CONVERTER. (4) ANALOG SUMMER.	ERROR SIGNAL TO MDA IS LIMITED JOINT WILL NOT DRIVE OR WILL DRIVE SLOWER. ARM MOTION MAY RESULT IN AN UNEXPECTED TRAJECTORY. WORST CASE ---------- UNEXPECTED MOTION. SLUGGISH JOINT. UNANNUNCIATED. CREW ACTION REQUIRED. REDUNDANT PATHS REMAINING ---------- N/A		QA/INSPECTIONS ---------- UNITS ARE MANUFACTURED UNDER DOCUMENTED QUALITY CONTROLS. THESE CONTROLS ARE EXERCISED THROUGHOUT DESIGN PROCUREMENT, PLANNING, RECEIVING, PROCESSING, FABRICATION, ASSEMBLY, TESTING AND SHIPPING OF THE UNITS. MANDATORY INSPECTION POINTS ARE EMPLOYED AT VARIOUS STAGES OF FABRICATION ASSEMBLY AND TEST. GOVERNMENT SOURCE INSPECTION IS INVOKED AT VARIOUS CONTROL LEVELS. EEE PARTS INSPECTION IS PERFORMED AS REQUIRED BY SPAR-RMS-PA.003. EACH EEE PART IS QUALIFIED AT THE PART LEVEL TO THE REQUIREMENTS OF THE APPLICABLE SPECIFICATION. ALL EEE PARTS ARE 100% SCREENED AND BURNED IN, AS A MINIMUM, AS REQUIRED BY SPAR-RMS-PA.003, BY THE SUPPLIER. ADDITIONALLY, EEE PARTS ARE 100% RE-SCREENED IN ACCORDANCE WITH REQUIREMENTS, BY AN INDEPENDENT SPAR APPROVED TESTING FACILITY. DPA IS PERFORMED AS REQUIRED BY PA.003 ON A RANDOMLY SELECTED 5% OF PARTS, MAXIMUM 5 PIECES, MINIMUM 3 PIECES FOR EACH LOT NUMBER/DATE CODE OF PARTS RECEIVED. WIRE IS PROCURED TO SPECIFICATION MIL-W-22759 OR MIL-W-81381 AND INSPECTED AND TESTED TO NASA JSCM08OD STANDARD NUMBER 95A. RECEIVING INSPECTION VERIFIES THAT ALL PARTS RECEIVED ARE AS IDENTIFIED IN THE PROCUREMENT DOCUMENTS, THAT NO PHYSICAL DAMAGE HAS OCCURRED TO PARTS DURING SHIPMENT, THAT THE RECEIVING DOCUMENTS PROVIDE ADEQUATE TRACEABILITY INFORMATION AND SCREENING DATA CLEARLY IDENTIFIES ACCEPTABLE PARTS. PARTS ARE INSPECTED THROUGHOUT MANUFACTURE AND ASSEMBLY AS APPROPRIATE TO THE MANUFACTURING STAGE COMPLETED. THESE INSPECTIONS INCLUDE, PRINTED CIRCUIT BOARD INSPECTION FOR TRACK SEPARATION, DAMAGE AND ADEQUACY OF PLATED THROUGH HOLES, COMPONENT MOUNTING INSPECTION FOR CORRECT SOLDERING, WIRE LOOPING, STRAPPING, ETC. OPERATORS AND INSPECTORS ARE TRAINED AND CERTIFIED TO NASA NHB 5300.4(3A) STANDARD, AS MODIFIED BY JSC 08800A. CONFORMAL COATING INSPECTION FOR ADEQUATE PROCESSING IS PERFORMED USING ULTRAVIOLET LIGHT TECHNIQUES. POST P.C. BD. INSTALLATION INSPECTION, CLEANLINESS AND WORKMANSHIP (SPAR/GOVERNMENT REP. MANDATORY INSPECTION POINT) P.C. BD. INSTALLATION INSPECTION, CHECK FOR CORRECT BOARD INSTALLATION, ALIGNMENT OF BOARDS, PROPER CONNECTOR CONTACT MATING, WIRE ROUTING, STRAPPING OF WIRES ETC., PRE-CLOSURE INSPECTION, WORKMANSHIP AND CLEANLINESS (SPAR/GOVERNMENT REP. - MANDATORY INSPECTION POINT) PRE-ACCEPTANCE TEST INSPECTION, WHICH INCLUDES AN AUDIT OF LOWER TIER INSPECTION COMPLETION, AS BUILT CONFIGURATION VERIFICATION TO AS DESIGN ETC., (MANDATORY INSPECTION POINT).

PREPARED BY: MEWG
SUPERCEDING DATE: 01 OCT 86
APPROVED

DATE:

CRITICAL ITEMS LIST

PROJECT: SRMS
ASS'Y NOMENCLATURE: SERVO POWER AMPLIFIER
SYSTEM: ELECTRICAL SUBSYSTEM
ASS'Y P/N: 2114071177
SHEET: 4

FMEA REF.	REV.	NAME QTY. & DRAWING REF. DESIGNATION	FAILURE MODE AND CAUSE	FAILURE EFFECT ON NEXT ITEM	MDUR / FUNC. I/F CRITICALITY	RATIONALE FOR ACCEPTANCE
2560	1	MOTOR CONTROL LOOP QTY-6 SCHEMATIC 2563721 AND 2563722	MODE: OUTPUT FROM ANALOG DRIVE SIGNAL TO MDA LOWER THAN COMMANDED. CAUSE(S): (1) INPUT SHIFT REGISTER AND LATCH. (2) DIGITAL COMPARATOR AND LIMITER. (3) D TO A CONVERTER. (4) ANALOG SUMMER.	ERROR SIGNAL TO MDA IS LIMITED JOINT WILL NOT DRIVE OR WILL DRIVE SLOWER. ARM MOTION MAY RESULT IN AN UNEXPECTED TRAJECTORY. WORST CASE ---------- UNEXPECTED MOTION, SLUGGISH JOINT. UNANNUNCIATED. CREW ACTION REQUIRED. REDUNDANT PATHS REMAINING ---------- N/A		A TEST READINESS REVIEW (TRR) WHICH INCLUDES VERIFICATION OF TEST PERSONNEL, TEST DOCUMENTS, TEST EQUIPMENT CALIBRATION/ VALIDATION STATUS AND HARDWARE CONFIGURATION IS CONVENED BY QUALITY ASSURANCE IN CONJUNCTION WITH ENGINEERING, RELIABILITY, CONFIGURATION CONTROL, SUPPLIER AS APPLICABLE, AND THE GOVERNMENT REPRESENTATIVE, PRIOR TO THE START OF ANY FORMAL TESTING (ACCEPTANCE OR QUALIFICATION). ACCEPTANCE TESTING (ATP) INCLUDES AMBIENT PERFORMANCE, THERMAL AND VIBRATION TESTING, (SPAR/GOVERNMENT REP. - MANDATORY INSPECTION POINT). INTEGRATION OF UNIT TO JOINT SRU - INSPECTIONS INCLUDE GROUNDING CHECKS, CONNECTORS FOR DENT OR PUSHBACK CONTACTS, VISUAL, CLEANLINESS, INTERCONNECT WIRING AND POWER UP TEST TO THE APPROPRIATE JOINT INSPECTION TEST PROCEDURE (TIP) ETC. JOINT LEVEL PRE-ACCEPTANCE TEST INSPECTION, INCLUDES AN AUDIT OF LOWER TIER INSPECTION COMPLETION, AS BUILT CONFIGURATION VERIFICATION TO AS DESIGN ETC. JOINT LEVEL ACCEPTANCE TESTING (ATP) INCLUDES AMBINET, VIBRATION AND THERMAL-VAC TESTING. (SPAR/GOVERNMENT REP. - MANDATORY INSPECTION POINT). SRMS SYSTEMS INTEGRATION, THE INTEGRATION OF MECHANICAL ARM SUBASSEMBLIES AND THE FLIGHT CABIN EQUIPMENT TO FORM THE SRMS. INSPECTIONS ARE PERFORMED AT EACH PHASE OF INTEGRATION WHICH INCLUDES GROUNDING CHECKS, THRU WIRING CHECKS, WIRING ROUTING, INTERFACE CONNECTORS FOR DENT OR PUSH BACK CONTACTS ETC. SRMS SYSTEMS TESTING - STRONGBACK AND FLAT FLOOR AMBIENT PERFORMANCE TEST. (SPAR/GOVERNMENT REP. - MANDATORY INSPECTION POINT)

PREPARED BY: RPLN
SUPERSEDING DATE: 01 OCT 88
APPROVED B
DATE:

CRITICAL ITEMS LIST

PROJECT: SRMS
ASS'Y NOMENCLATURE: SERVO POWER AMPLIFIER
SYSTEM: ELECTRICAL SUBSYSTEM
ASS'Y P/N: 5114OF1177
SHEET: 5

FMEA REF.	REV.	NAME QTY & DRAWING REF. DESIGNATION	FAILURE MODE AND CAUSE	FAILURE EFFECT ON END ITEM	HDWR / FUNC. I/I CRITICALITY	RATIONALE FOR ACCEPTANCE
2560	2	MOTOR CONTROL LOOP QTY-6 SCHEMATIC 2563721 AND 2563722	MODE: OUTPUT FROM ANALOG DRIVE SIGNAL TO MDA LOWER THAN COMMANDED. CAUSE(S): (1) INPUT SHIFT REGISTER AND LATCH. (2) DIGITAL COMPARATER AND LIMITER. (3) D TO A CONVERTER. (4) ANALOG SUMMER.	ERROR SIGNAL TO MDA IS LIMITED JOINT WILL NOT DRIVE OR WILL DRIVE SLOWER. ARM MOTION MAY RESULT IN AN UNEXPECTED TRAJECTORY. WORST CASE --- UNEXPECTED MOTION. SLUGGISH JOINT. UNANNUNCIATED. CREW ACTION REQUIRED. REDUNDANT PATHS REMAINING --- N/A		FAILURE HISTORY --- THE FOLLOWING FAILURE ANALYSIS REPORT(S) ARE RELEVANT: FAR 3014: S/N 209 NOV 78 DESCRIPTION --- MDA2 TEST ERRORS, DEMAND 776, SUMMER RATE OUTPUT -10.4V, LOWER LIMIT 0.07V. CORRECTIVE ACTION --- NO CORRECTIVE ACTION, POSSIBLE SOFTWARE FAULT. FAR 3330: S/N 206,207 MAY 87 DESCRIPTION --- ANALOG TACHOMETER GAIN LOW, S.O.T. RESISTOR INADEQUATE. CORRECTIVE ACTION --- MODIFY TP2573445 TO IMPROVE S.O.T.

PREPARED BY: MFWG
SUPERCEDING DATE: 06 OCT 87
APPROVED
DATE:

CRITICAL ITEMS LIST

PROJECT: SAMS
ASS'Y NOMENCLATURE: SERVO POWER AMPLIFIER
SYSTEM: ELECTRICAL SUBSYSTEM
ASS'Y P/N: 51A0T1??
SHEET: 6

FMEA REF	REV	NAME, QTY, & DRAWING REF DESIGNATION	FAILURE MODE AND CAUSE	FAILURE EFFECT ON END ITEM	HDWR / FUNC. I/I CRITICALITY	RATIONALE FOR ACCEPTANCE
2560	1	ROTOR CONTROL LOOP QTY 6 SCHEMATIC 2561721 AND 2561722	MODE: OUTPUT ERROR ANALOG DRIVE SIGNAL TO MDA LOWER THAN COMMANDED CAUSE(S): (1) INPUT SHIFT REGISTER AND LATCH. (2) DIGITAL COMPARATOR AND LIMITER. (3) D TO A CONVERTER. (4) ANALOG SUMMER.	ERROR SIGNAL TO MDA IS LIMITED JOINT WILL NOT DRIVE OR WILL DRIVE SLOWER. ARM MOTION MAY RESULT IN AN UNEXPECTED TRAJECTORY WORST CASE UNEXPECTED MOTION. SLUGGISH JOINT. UNANNUNCIATED. CREW ACTION REQUIRED. REDUNDANT PATHS REMAINING N/A	1	OPERATIONAL EFFECTS ARM DOES NOT RESPOND PROPERLY TO HAND CONTROLLER COMMANDS OR AUTO SEQUENCES. CREW INHERENTLY COMPENSATES FOR ANY UNDESIRED ARM TRAJECTORY IN MANUAL AUGMENTED MODES. CREW ACTION APPLY BRAKES. SELECT BACKUP. CREW TRAINING THE CREW WILL BE TRAINED TO OBSERVE WHETHER THE ARM IS RESPONDING PROPERLY TO COMMANDS. IF IT ISN'T, APPLY BRAKES. MISSION CONSTRAINT OPERATE UNDER VERNIER RATES WITHIN 10 FT OF STRUCTURE. THE OPERATOR MUST BE ABLE TO DETECT THAT THE ARM IS RESPONDING PROPERLY TO COMMANDS VIA WINDOW AND/OR CCTV VIEWS DURING ALL ARM OPERATIONS. AUTO TRAJECTORIES MUST BE DESIGNED TO COME NO CLOSER THAN 5 FT FROM STRUCTURE. SCREEN FAILURES N/A OMRSD OFFLINE OPERATE EACH JOINT IN COMPUTER CONTROLLED MODE VERIFY THAT ACTUAL RATES AGREE WITH COMMANDED OMRSD ONLINE INSTALLATION NONE OMRSD ONLINE TURNAROUND NONE

PREPARED BY: MFWG
SUPERCEDING DATE: 21 OCT 87
APPROVED BY:
DATE: