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EMU CRITICAL ITEMS LIST 5/30/2002 SHDERSEDES

EMU CRITICAL ITEMS LIST			5/30/200 12/31/20	2 SUPERSEDES 01	Date: 6/5/2002
NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE	
JUMPER HARNESS, ITEM 391	2/2	Electrical short to ground in	END ITEM: Electrical short in	A. Design - Short circuits are minimized by the following: Each connector/adapter ring interface is locked in	place to prevent rotation
SV821755-1 (1)		warning tone or status tone lines.  Cable chafing against connector shell or shield.  Improper connector strain relief, insulation breakdown.	warning tone or status tone lines to ground.	a mechanical lock. #24 AWG Teflon insulated wires as electrical conduction and insulation properties. Coat 56.7% of derated temperature and 4.3% of derated than 1% of derated current. The convoluted tubing prof insulation to prevent shorts between the EMI braunshielded conductors. The woven Halar sheath is as	nd connector provide onnector pins are operat voltage, and wire is at rovides an additional laid and any internal
			GFE INTERFACE: Tones will be continuously activated.  MISSION: Terminate EVA	cables to provide protection from abrasion and impainsulated by a polyphenylene sulfide insert. The P3 has internal edges blended smooth to prevent cable provided by the combination of convolute tubing, mextra cable length. The braided items are secured by connector/cable interface. The convolute tubing is Wire crimping is performed per SVHS4909 (based on M	ct. Connector pins are connector backshell how chafing. Strain relief is etal EMI braid, and 0.5' y a band strap at each threaded into the connect
			due to crew discomfort and loss of warning tones for future warning messages.  CREW/VEHICLE:	B. Test - Component Acceptance Test - The 391 harness is subjected to acceptance testing per AT-E-391 acceptance to ensure there are no workmanship problems that coul or short circuit. Each connector/harness interface is subjected test. The insulation resistance between each conductor and the is measured during this test to ensure there are no intermittent verify the integrity of the harness strain relief. A continuity performed to measure the resistance of each circuit to ensure th	ems that could cause an is subjected to a 9-lb. ctor and the ground circ intermittent shorts and continuity test is to ensure there are no
			None.  TIME TO EFFECT /ACTIONS: Seconds.	circuits or high resistance paths. The insulation restrength between each conductor and the shield grown there are no shorts.  PDA Test - The warning tone and status tone lines are checked of SEMU-60-015 para. 4.0 (Electrical Testing).	nd is measured to ensure
			TIME AVAILABLE: N/A	Certification Test - Certified for a useful life of 15 years (ref. EMU1-	13-046).
			TIME REQUIRED: N/A	C. Inspection - To ensure that there are no workmanship problems who circuit in the harness conductors, the following in	spections are made: Cont
			REDUNDANCY SCREENS: A-N/A B-N/A C-N/A	crimp samples are made prior to start of crimping as crimping and pull tested to ensure the crimp tooling crimp terminations are inspected for defects. Harner visually inspected prior to assembly to ensure there cause a short due to workmanship. Electrical bond to ground path through various points on the harness.	g is operating properly. ss cables and conductors e are no defects which c est is performed to veri In-process and final

D. Failure History -None.

circuits.

electrical checkout of the harness (conductor continuity, dielectric streng and insulation resistance tests) are performed to ensure there are no open

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391FM03

			12/31/200	, <u> </u>	
NAME		FAILURE			
P/N		MODE &			
QTY	CRIT	CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE	

E. Ground Turnaround -

Ground Turnaround tested per FEMU-R-001, Tones Test.

#### F. Operational Use -

Crew Response -PreEVA: Trouble shoot problem. Consider third EMU if availateminate EVA prep due to crew discomfort caused by continuous tone and los EMU annunciation capability. EVA: Terminate EVA. EMU is go for SCU if noise level is bearable.

Date: 6/5/2002

Training - Standard training covers this failure mode.

#### Operational Considerations -

Flight rule Al5.1.2-2 of "Space Shuttle Operational Flight Rules", NSTS-128 defines go/no go criteria related to EMU CWS. Generic EVA Checklist, JSC-4 procedures Section 3 (EMU Checkout) and 4 (EVA prep) verify hardware integrand systems operational status prior to EVA. Real Time Data System allows ground monitoring of EMU systems.

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C-PASS

Date: 6/5/2002 NAME FAILURE P/N MODE & CRIT CAUSES OTY FAILURE EFFECT RATIONALE FOR ACCEPTANCE 391FM04 JUMPER HARNESS, 2/1R Electrical END ITEM: A. Design -ITEM 391 short in Fan Short circuit Short circuits are minimized by the following: Each connector/adapter ring Switch Line. across battery interface is locked in place to prevent rotation by a mechanical lock. #22 SV821755-1 Teflon insulated wires and connector provide electrical conduction and insulation properties. Connector pins are operating at 56.7% of derated (1) Cable chafing GFE INTERFACE: temperature and 6.5% of derated voltage, and wire is at 4.4% of derated cur Loss of power to PLSS (no fan, no against The convoluted tubing provides an additional layer of insulation to prevent shorts between the EMI braid and any internal unshielded conductors. The wo connector shell or Halar sheath is assembled over the internal cables to provide protection fi shield. communications abrasion and impact. Connector pins are insulated by a polyphenylene sulfic improper . Current connector meter shunt insert. The P3 connector backshell housing has internal edges blended smooth prevent cable chafing. Strain relief is provided by the combination of strain relief, will fuse open convolute tubing, metal EMI braid, and 0.5 in. extra cable length. The braid insulation items are secured by a band strap at each connector/cable interface. The breakdown. MISSION: convolute tubing is threaded into the connectors. Wire crimping is performe Loss of one SVHS4909 (based on MSFC Spec-Q-1A). EMU. Terminate EVA. B. Test -Component Acceptance Test -The 391 harness is subjected to acceptance testing per AT-E-391 prior to fi CREW/VEHICLE: acceptance to ensure there are no workmanship problems that could cause an None for or short circuit. Each connector/harness interface is subjected to a 9-lb. single test. The insulation resistance between each conductor and the ground circ failure. is measured during this test to ensure there are no intermittent shorts and Possible crew verify the integrity of the harness strain relief. A continuity test is loss with loss performed to measure the resistance of each circuit to ensure there are no of SOP. circuits or high resistance paths. The insulation resistance and dielectri strength between each conductor and the shield ground is measured to ensure TIME TO EFFECT there are no shorts. /ACTIONS: Seconds. PDA Test -The fan switch lines are checked during the DCM PDA SEMU-60-015 para. 4.0 TTMF (Electrical Testing). AVAILABLE: Minutes. Certification Test -Certified for a useful life of 15 years (ref. EMU1-13-046). TIME REOUIRED: Seconds. C. Inspection -To ensure that there are no workmanship problems which could cause a short REDUNDANCY circuit in the harness conductors, the following inspections are made: Cont SCREENS: crimp samples are made prior to start of crimping and at the conclusion of A-PASS crimping and pull tested to ensure the crimp tooling is operating properly B-PASS crimp terminations are inspected for defects. Harness cables and conductor

> D. Failure History -None.

circuits.

visually inspected prior to assembly to ensure there are no defects which ( cause a short due to workmanship. Electrical bond test is performed to veri ground path through various points on the harness. In-process and final electrical checkout of the harness (conductor continuity, dielectric streng and insulation resistance tests) are performed to ensure there are no open,

# EXTRAVEHICULAR MOBILITY UNIT

### SYSTEMS SAFETY REVIEW PANEL REVIEW

FOR THE

# I-391 JUMPER POWER HARNESS

CRITICAL ITEM LIST (CIL)

EMU CONTRACT NO. NAS 9-97150

Prepared by: Approved by: Approved by: NASA - SSM SSM

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