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EMU CRITICAL ITE	MS LIST		5/30/200 12/31/20	2 SUPERSEDES 01	Date: 6/5/2002
NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE	
JUMPER HARNESS, ITEM 391	2/2	Electrical open or short in feedwater	END ITEM: Electrical open or short	A. Design - Open and short circuits are minimized by the followi Each connector/adapter ring interface is locked in p	
SV821755-1 (1)		open/close lines.	to ground in feedwater open/close lines. These	a mechanical lock. #22 AWG Teflon insulated wires a electrical conduction and insulation properties. Coderated temperature and 6.9% of derated voltage, and current. The convoluted tubing provides an additional	nd connector provide nnector pins are at 56. wire is at 12.4% of de
		Cable chafing against connector shell or shield.	lines are current limited to 0.7 +/- 0.2 amps in the DCM.	prevent shorts between the EMI braid and any internation woven Halar sheath is assembled over the internal cafrom abrasion and impact. Connector pins are insulat sulfide insert. The P3 connector backshell housing his smooth to prevent cable chafing. Strain relief is preconvolute tubing, metal EMI braid, and 0.5" extra ca	bles to provide protect ed by a polyphenylene as internal edges blend ovided by the combinati
		connector strain relief. Faulty connection	GFE INTERFACE: Loss of power to 137 valve.	items are secured by a band strap at each connector/convolute tubing is threaded into the connectors. Wi SVHS4909 (based on MSFC Spec-Q-1A).	cable interface. The
		between the	Unable to	B. Test -	
		connector and the lead wires, insulation	change position of feedwater valve (137).	Component Acceptance Test - The 391 harness is subjected to acceptance testing pacceptance to ensure there are no workmanship proble or short circuit. Each connector/harness interface i	ms that could cause an s subjected to a 9-lb.
		breakdown, conductor severed, contact resistance.	If closed, no cooling or dehumidificati n available.	test. The insulation resistance between each conduction is measured during this test to ensure there are no verify the integrity of the harness strain relief. A performed to measure the resistance of each circuit circuits or high resistance paths. The insulation re	intermittent shorts and continuity test is to ensure there are no
			MISSION: Terminate EVA. Loss of EMU.	strength between each conductor and the shield groun there are no shorts.	
			CREW/VEHICLE:	PDA Test - The feed-water open/close lines are checked during D 015 para. 4.0 (Electrical Testing).	CM PDA testing per SEMU
			None for single failure	Certification Test - Certified for a useful life of 15 years (ref. EMU1-1	3-046).
			TIME TO EFFECT	C. Inspection -	
			/ACTIONS: Seconds.	To ensure that there are no workmanship problems whi short circuit in the harness conductors, the followi Contact crimp samples are made prior to start of cri	ng inspections are made
			TIME AVAILABLE: N/A	conclusion of crimping and pull tested to ensure the properly. All crimp terminations are inspected for conductors are visually inspected prior to assembly defects which could cause an open or short due to wo	crimp tooling is operadefects. Harness cableato ensure there are no
			TIME REQUIRED:	defects which could cause an open or short due to wo test is performed to verify ground path through vari	ous points on the harne

SCREENS: A-N/AD. Failure History -B-N/A None.

N/A

C-N/A

REDUNDANCY

In-process and final electrical checkout of the harness (conductor

continuity, dielectric strength, and insulation resistance tests) are perfeto ensure there are no open/short circuits.

EMU CRITICAL ITEMS LIST 5/30/2002 SUPERSEDES

FAILURE

CTT.

NAME

12/31/2001 Date: 6/5/2002

E. Ground Turnaround -

Tested per FEMU-R-001, Item 137 Feedwater Shutoff Valve Functional Verifica

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F. Operational Use -

Crew Response $\,$ -PreEVA: Trouble shoot problem. Consider third EMU if availating If no success continue EVA prep.

EVA: When CWS data confirms loss of sublimator pressure due to failed close valve and cooling is not adequate, terminate EVA. When failed open valve is detected by water in airlock during repress, perform water dump of feedwate tanks.

Training - Standard training covers this failure mode.

Operational Considerations -

Flight rule A15.1.2-2 of "Space Shuttle Operational Flight Rules", NSTS-128 defines go/no go criteria related to EMU thermal control. Generic EVA Checklist, JSC-48023, procedures Section 3 (EMU Checkout) and 4 (EVA prep) verify hardware integrity and systems operational status prior to EVA. Realime Data System allows ground monitoring of EMU systems.

EXTRAVEHICULAR MOBILITY UNIT

SYSTEMS SAFETY REVIEW PANEL REVIEW

FOR THE

I-391 JUMPER POWER HARNESS

CRITICAL ITEM LIST (CIL)

EMU CONTRACT NO. NAS 9-97150

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