CIL

FMI CPITICAL ITEMS LIST

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EMU CRITICAL ITE	MS LIST		5/30/2002 12/31/200	2 SUPERSEDES	Page 1 Date: 6/5/2002
NAME P/N QTY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE	
		392FM05	. – – – – – –		
JUMPER HARNESS, ITEM 392	2/1RB	Electrical open, warning tone or status	END ITEM: Loss of continuity in	A. Design - Open circuits are minimized by the following: Each connector/adapter ring interface is locked in place to prevent rotation by a mechanical lock. #2	
SV821756-2 (1)		Cable chafing against connector shell or shield. Improper connector strain relief. Faulty connection between the connector and the lead wires, conductor severed, contact resistance.	warning tone or status tone line. GFE INTERFACE: No audible tones when activated by CWS. MISSION: None for single failure. Crew would not be alerted to subsequent failures and could not properly respond with corrective action. CREW/VEHICLE: None for single failure. Possible loss of crewman with loss of CCC, oxygen or low vent flow. TIME TO EFFECT /ACTIONS: Minutes. TIME AVAILABLE: Minutes. TIME REQUIRED: Minutes.	Teflon insulated wires and connector provide electrica insulation properties to prevent wire breakage and to Connector pins are operating at 56.7% of derated tempe of derated current. The woven Halar sheath is assemble to provide protection from abrasion and impact. Strain combination of convolute tubing, metal EMI braid, and The braided items are secured by a band strap at each The convolute tubing is threaded into the connectors. per SVHS4909 (based on MSFC Spec-Q-1A). B. Test - Component Acceptance Test - The 392 harness is subjected to acceptance testing per acceptance to ensure there are no workmanship problems or short circuit. Each connector/harness interface is test. The insulation resistance between each conductor is measured during this test to ensure there are no in verify the integrity of the harness strain relief. A c performed to measure the resistance of each circuit to circuits or high resistance paths. The insulation resistength between each conductor and the shield ground there are no shorts. PDA Test - The warning tone and status lines are checked during D 015 para. 4.0 (Electrical Testing). Certification Test - Certified for a useful life of 15 years (ref. EMUI-13-C. Inspection - To ensure that there are no workmanship problems which circuit in the harness conductors, the following inspecting samples are made prior to start of crimping and crimping and pull tested to ensure the crimp tooling ic rimp terminations are inspected for defects. Harness visually inspected prior to assembly to ensure there a cause an open due to workmanship. Electrical bond tes ground path through various points on the harness. Inelectrical checkout of the harness (conductor continui and insulation resistance tests) are performed to ensurincuits. D. Failure History - None. E. Ground Turnaround - Tested per FEMU-R-001, Tones Test.	l conduction and help prevent shorting rature and wire at 14 dover the internal carelief is provided by the connector/cable interwire crimping is performed to a 9-1b. AT-E-392 prior to finant that could cause an subjected to a 9-1b. It and the ground circular termittent shorts and continuity test is ensure there are no stance and dielectrical is measured to ensure the conclusion of soperating properly. CM PDA testing per SI O46). Could cause an open cations are made: Corat the conclusion of soperating properly. Cables and conductors re no defects which cat is performed to very process and final ty, dielectric streng

REDUNDANCY F. Operational Use -

EMU CRITICAL ITEMS LIST 5/30/200

FAILURE

CIL

NAME

5/30/2002 SUPERSEDES 12/31/2001

P/N QTY	CRIT	MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE

SCREENS: Crew Response A-PASS PreEVA: Troubleshoot, if no success, consider third EMU if available.
B-FAIL Otherwise, EMU go for EVA. Rely on visual monitoring of displayed messages.
C-PASS If detected during airlock depress, continue EVA. Rely on visual monitoring of displayed messages.

Training - Standard EMU training covers this failure mode.

Operational Considerations -

Flight rule A15.1.2-2 of "Space Shuttle Operational Flight Rules", NSTS-128 defines go/no go criteria related to EMU CWS. Generic EVA Checklist, JSC-4 procedures Section 3 (EMU Checkout) and 4 (EVA prep) verify hardware integrand systems operational status prior to EVA. Real Time Data System allows ground monitoring of EMU systems.

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Date: 6/5/2002

EXTRAVEHICULAR MOBILITY UNIT

SYSTEMS SAFETY REVIEW PANEL REVIEW

FOR THE

I-392 JUMPER SIGNAL HARNESS

CRITICAL ITEM LIST (CIL)

EMU CONTRACT NO. NAS 9-97150

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