| CIL<br>EMU CRITICAL ITEMS LIST                |                  | 5/30/200<br>12/31/20  | 2 SUPERSEDES<br>01  | Page 1<br>Date: 6/5/2002  |
|---|------------------|---|---|---|
| NAME  | FAILURE          |   |   |   |
| P/N<br>QTY CRIT                               | MODE &<br>CAUSES | FAILURE EFFECT  | RATIONALE FOR ACCEPTANCE  |   |
|   | 490FM03z         |   |   |   |
| BATTERY, ITEM 490 2/1R4<br>SV819600-01<br>(1) |                  | END ITEM:<br>Continuous<br>electrolyte<br>flow through<br>the relief<br>valve.<br>GFE INTERFACE:<br>If relief<br>valve fails<br>open, the<br>electrolyte<br>would<br>sublimate to<br>vacuum and<br>battery power<br>would be lost.<br>MISSION:<br>Terminate EVA<br>with loss of<br>battery power.<br>CREW/VEHICLE:<br>None for<br>single<br>failure.<br>Possible loss<br>of crewmember<br>with loss of<br>SOP.<br>TIME TO EFFECT<br>/ACTIONS:<br>Seconds.<br>TIME<br>AVAILABLE:<br>Minutes.<br>TIME REQUIRED:<br>Seconds.<br>REDUNDANCY<br>SCREENS:<br>A-FAIL<br>B-PASS<br>C-PASS | <ul> <li>A. Design - The valve seal is a quad "Q" seal made from ethylene pro in the poppet head and seals against the valve housing. a 32 micro-inch finish for smoothness. Contamination is the valve by a 304 CRES sintered wire 0.030" mesh screene each relief valve. The screens are held in place by Tef poppet spring is made from 17-7 PH stainless steel with 155.00 ksi. At the maximum operating pressure of 40 psi stressed to 103.517 ksi. The spring has a yield strengt B. Test - The relief valve is tested for reseat per AT-E-490-RV-1. is 31 psid minimum.</li> <li>PDA Test - Data is transferred from the Acceptance Test.</li> <li>Certification Test - A relief valve was cycled through crack, flow, and resea 1999. All reseat pressures recorded were above the mini specification of 28 psig. The item has completed all ce C. Inspection - The relief valves are tested and adjusted during vendor that the valves reseat at the specified 31 psid minimum.</li> <li>D. Failure History - B-EMU-490-A027 (2/23/01) Relief valves failed reseat requirements of 28 - 28.1 ps during crack and reset screening. The cause was due for shift in the valve set point, which was the result of th partly to stiction of the Quad seals. BRV (battery reli to eliminate set point shift and stabilize the spring by SV819600/2 drawing was also revised to update note 8 to crack flow rate and as a maximum reseat flow rate.</li> <li>H-EMU-490002 (1/23/01) Battery relief valve failed visual inspection when the H calibration nut became partially debonded. This problem mixing of the Hysol leading to the improper cure for Hys Krytox was found on the nut surface and poppet thread. prevented adhesion of Hysol. Battery relief valve was r Hysol and incorporate a new primary calibration nut rete secondary nut retention feature.</li> <li>E. Ground Turnaround - None. Due to battery design, no ground turnaround test F. Operational Use - Crew Response - PreEVA/PostEVA: Swap out EMU battery. EVA: When loss of battery power detected, terminate EVA</li> </ul> | The sealing surface has<br>prevented from entering<br>installed directly over<br>lon caps. The valve<br>a yield strength of<br>g., the spring is<br>h safety factor of 1.5.<br>Reseat of assemblies<br>t for 148 cycles in Jur<br>mum required<br>rtification requirement<br>manufacturing to verify<br>ig with 0 cc/min flow<br>the most part to the<br>e spring movement and<br>ef valve) was redesigned<br>adding a shim. S/AD<br>include 5 cc as minimum<br>ysol used to retain the<br>was caused by poor<br>ol. In addition,<br>This lubricant material<br>edesigned to eliminate<br>ntion feature and add a |

| CIL<br>EMU CRITICAI | L ITEMS LIST |                             | 5/30/2002<br>12/31/200 | 2 SUPERSEDES Date: 6/5/2002   | _ |
|---------------------|--------------|-----------------------------|------------------------|---|---|
| NAME<br>P/N<br>QTY  | CRIT         | FAILURE<br>MODE &<br>CAUSES | FAILURE EFFECT         | RATIONALE FOR ACCEPTANCE  |   |
|                     |              | 490FM03Z                    |                        | Training: Standard EMU training covers this failure mode.   | - |
|                     |              |                             |                        | Operational Considerations -<br>Flight rule A15.1.2-2 of "Space Shuttle Operational Flight Rules", NSTS-12820 |   |

defines go/no go criteria related to EMU battery. Generic EVA Checklist, JSC-48023, procedures Section 3 (EMU Checkout) and 4 (EVA prep) verify hardware integrity and systems operational status prior to EVA. Real Time Data System allows ground monitoring of EMU systems.

# EXTRAVEHICULAR MOBILITY UNIT

#### SYSTEMS SAFETY REVIEW PANEL REVIEW

## FOR THE

## **I-490 BATTERY**

# CRITICAL ITEM LIST (CIL)

### EMU CONTRACT NO. NAS 9-97150

<u>M. Snyln</u> HS - Reliability

<u>R. Munfard 4/24/02</u> HS - Engineering Manager

ABJanco S/21/0

S&MA stado

5-30-02

Joe Taun 6/04/02 NASA - Crew Jac Manager NASA - Program Manager