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CEL
CRITICAL ITEMS LIST

ASSEMBLY NAME/PART NUMBER: LATCH PIN CADDY/10154-20292-01
 Reference: (PERDEFI)
 Prepared By: C. Hartman Approved By: R. Withey
 Superseding Date: 8/80 Date: 1/81 Revs: A

NAME P/N QUANTITY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
Latch Pin Caddy 11400-90040-02 Item 5.1 One	III	5.17005 Loss of Latch Ring. CAUSE: Defective material. Impact. Loss of interface adapter bracket screws.	END ITEM: Caddy separates from latch. Loss of Caddy. RFE INTERFACE: Loss of Latch Pin Assemblies. MISSION: Unable to secure latches. Terminate EVA. CREW/VEHICLE: Vehicle damaged by loose latches during ejection. Loss of crew and vehicle.	A. DESIGN: The Latch Pin Caddy Latch Ring is fabricated from 4041-1651 aluminum and is anodized according to MIL-A-8629, Type III, Class I 100-A-250 101. The interface adapter bracket and latch ring are installed with a set of three screws fabricated from stainless steel and procured to MS specification. Loss of screws is precluded in design by adherence to standard engineering torque requirements for screw installation. The screws are installed and torqued to 20 in/lbs to ensure that they remain in place. The Latch Pin Caddy is stored in a foam cushion in a Payload Bay PSA to protect it from the possibility of impact from the possibility of damage from impact. B. TEST: Component Acceptance Test None PBA Test - The following tests are conducted at the Latch Pin Caddy assembly level in accordance with ILC Document 10107-70690:

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ECR
CAPITAL ITEMS LIST

ASSEMBLY NAME/PART NUMBER: LATCH PIN CADDY/64-159-24292 #1
Reference: IFCACRIL
Prepared By: C. Harlan
Approved By: N. Willey
Superseding Bales: B700
Date: 1189 Rev: A

NAME P/N QUANTITY	CRG	FAILURE MODE & CAUSES	FAILURE EFFECT	REVISIONS FOR ACCEPTANCE
Latch Pin Caddy 64101-10000-02 Ilse S.1 One	178	S.1FN05 Loss of lather ring.		1. Interface with wrist tether hook. Certification test - None C. INSPECTION: Components and material manufactured to IIC requirements at an approved supplier are documented from procurement through shipping by the supplier. IIC incoming receiving inspection verifies that the materials received are as idealized in the procurement documents, that no damage has occurred during shipment and that supplier certification has been received which provides traceability information. The following RIP's are performed during the lather ring manufacturing process to assure the failure causes are precluded from the fabricated items. 1. Visual inspection for damage or material degradation. 2. Verify presence of screws. 3. Witness screw torque.

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CRITICAL ITEMS (153)

ASSEMBLY NAME/PART NUMBER: LATCH PIN CADDY/10150 20242-01
 Reference: IPCAD7L
 Prepared By: E. Marlowe
 Superceding Dates: 0/00
 Approved By: R. Wilkey
 Date: 1/87 Rev. A

NAME P/N QUANTITY	COST	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
Latch Pin Caddy 01010-10004 02 Rev 5.1 One	170	5.100'S Loss of Latch Ring.		<p>During PDR, the following inspection points are performed at the Latch Pin Caddy Assembly level in accordance with ILC Document 18107-70870:</p> <ol style="list-style-type: none"> 1. Visual inspection for damage. 2. Verification of conformance to drawing. <p>D. FAILURE HISTORY: None</p> <p>E. GROUND TURNAROUNDS: During ground turnarounds, in accordance with ILC Document 18107-70711, the Latch Ring is interlocked with Latch Hook and visually inspected for damage.</p> <p>F. OPERATIONAL USE:</p> <ol style="list-style-type: none"> 1. Crew Response PRE/POST EVA - N/A EVA - If possible, attach Caddy to HMS to prevent loss. If Caddy is lost, attempt to remove both unpowered Satches from Payload Bay rails and transport to crew compartment for recovery stowage.

ELL
CRITICAL ITEMS LIST

ASSEMBLY NAME/PART NUMBER: LATCH PIN CAD00710104-00292-01
Reference: LPCADELL
Prepared By: C. Hartman
Supervising Date: 8/88

Approved By: M. Mithey
Date: 1/89 Rev: A

NAME P/N QUANTITY	CRIT	FAILURE MODE & CAUSES	FAILURE EFFECT	RATIONALE FOR ACCEPTANCE
Latch Pin Eaddy 70101-10-04-02 Item 5.1 One	1/1	S. ITEMS Loss of Retain Ring.		<ol style="list-style-type: none"> 1. Training Team trained in generic EVA Ops. 2. Operational Considerations Minimal impact. Tool usefulness unaffected. Task may require additional ties.

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